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Rate-dependent phase-field damage modeling of rubber and its experimental parameter identification

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Abstract

Phase-field models have the advantage in that no geometric descriptions of cracks are required, which means that crack coalescence and branching can be treated without additional effort. Miehe et al. [1] introduced a rate-independent phase-field damage model for finite strains in which a viscous damage regularization was proposed. We extend the model to depend on the loading rate and time by incorporating rubber's strain-rate dependency in the constitutive description of the bulk, as well as in the damage driving force. The parameters of the model are identified using experiments at different strain rates. Local strain fields near the crack tip, obtained with digital image correlation (DIC), are used to help identify the length scale parameter. Three different degradation functions are assessed for their accuracy to model the rubber's rate-dependent fracture. An adaptive time-stepping approach with a corrector scheme is furthermore employed to increase the computational efficiency with a factor of six, whereas an active set method guarantees the irreversibility of damage. Results detailing the energy storage and dissipation of the different model constituents are included, as well as validation results that show promising capabilities of rate-dependent phase-field modeling.

Keywords: Phase-field, fracture, damage, rubber, rate-dependent

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1 1. Introduction

2 Rubber products like seals, hoses and tires are widely used in industrial appli-
3 cations. In order to reduce the cost and time constraints to produce physical
4 prototypes, virtual prototypes can be developed instead. However, virtual pro-
5 totypes require adequate numerical simulation tools to describe the mechanical
6 responses. Several researchers have modeled the failure and fracture of rubber
7 materials [1] [2] [3], but rubber's rate-dependency is still relatively scarcely ad-
8 dressed.

9 Although [4], [5] and [6] have recognized the viscoelastic behavior as a major
10 factor affecting the crack growth rate, to the best of the authors' knowledge,
11 only [7] and [8] have incorporated it in predictive models with a node splitting
12 algorithm and a cohesive zone approach, respectively. These approaches have a
13 disadvantage in that they need either frequent remeshing or a-priori knowledge
14 of the crack path.

15 Phase-field damage models for fracture [9], also called variational approaches
16 to fracture [10], are recently gaining interest since they naturally manage crack
17 propagation, branching and coalescence without a-priori knowledge of the crack
18 path. This is achieved by treating the sharp discontinuity in a continuous man-
19 ner with a finite damage zone that is governed by a length scale parameter.
20 The similarities to gradient-enhanced damage models [11] [12] [13] [14] [15] are
21 obvious and highlighted in [16] and [17].

22 [1] was, according to the best of the authors' knowledge, the first to introduce
23 phase-field modeling for fracture of rubbery polymers. While already including
24 rate-dependency in the damage evolution, its aim was to add numerical stability
25 to the framework. An extension to anisotropic, hyperelastic materials, like soft
26 biological tissues, was presented in [18] and [19]. [20] and [21] used a phase-field
27 damage model to investigate the failure at the microscale of carbon black re-
28 inforced rubber composites. These works highlighted the ability of phase-field
29 damage approaches to model nucleation and coalescence of several cracks. The
30 fracture of silicone elastomers was studied in [2], whereas [3] introduced a mi-

31 cromechanically motivated definition of the crack driving force in a phase-field
32 approach for loosely crosslinked rubbery polymers. Evaluating various failure
33 criteria for soft biological materials, [22] favored a strain-energy based criterion
34 to describe damage evolution, which we use as well.

35 Phase-field approaches for fracture need a correct identification of the length
36 scale parameter. It can be shown that for an infinitesimally small length scale
37 the approach converges to a sharp crack surface [9]. This could lead to the
38 assumption that the length scale is a numerical parameter, which just needs to
39 be selected small enough. However, a substantial influence of the length scale
40 on the results is observed in [23]. Therefore we assume, as in [24], [25], [26], and
41 [27], that the length scale is a material parameter depending on the microstruc-
42 ture and needs to be calibrated with experimental data.

43
44 Although the general aim of this work is to develop a phase-field model to de-
45 scribe the rate-dependent failure of rubbers, four sub-aims can be distinguished.
46 First, we extend the proposed model of [1] to incorporate rate-dependency in
47 the bulk response as well as in the damage evolution. Second, since enforcing
48 the irreversibility of the damage field by the application of a local history field
49 [9] yields erroneous results for the rate-dependent formulation, we propose to
50 directly use the constraints on the evolution of the damage field. Third, we
51 introduce an adaptive time-stepping algorithm and use the corrector scheme of
52 [28] to reduce computation times. Fourth, [29] showed for a gradient-enhanced
53 damage model, that measurements of local strains near the crack tip are required
54 to correctly calibrate the fracture parameters, especially the length scale. We
55 experimentally identify all material parameters, including the length scale, such
56 that the computations for the presented validation tests are true predictions.

57 The paper is organized as follows. In chapter 2, the rate-dependent phase-field
58 damage model for finite strains is derived from energy-conservation. In chapter
59 3, we formulate the weak form, linearize and discretize our model. Special
60 attention is paid to the treatment of the irreversibility constraint of the damage
61 field. Chapter 4 presents the conducted experiments, the procedure to identify

62 the model's parameters and the validation results. We discuss amongst other
 63 things the value of the length scale parameter for which we have used digital
 64 image correlation (DIC) measurements and assess three different degradation
 65 functions in terms of accuracy to predict failure of rubber. The validation
 66 is performed by varying the specimen geometries and clamp velocities. We
 67 conclude this contribution in chapter 5.

68 In this work, we denote scalars by lowercase and capital letters (a and A),
 69 vectors by bold, lowercase letters (\mathbf{a}), second-order tensors by bold capitals (\mathbf{A})
 70 and fourth-order tensors by bold, capital italic letters (\mathcal{A}).

71 2. Energy-based rate-dependent phase-field damage model

72 In this section, the rate-dependent phase-field damage model is derived. We
 73 start by defining the kinematics. We consider a body Ω_0 in the reference configu-
 74 ration, with its external boundary denoted by $\partial\Omega_0$ and an internal discontinuity
 75 Γ_0 . The motion and deformation of the body are described by displacement \mathbf{u} ,
 76 deformation gradient $\mathbf{F} = \mathbf{I} + \nabla_0 \mathbf{u}$ and Green's strain tensor $\mathbf{E} = 1/2(\mathbf{F}^T \cdot \mathbf{F} - \mathbf{I})$.
 77 \mathbf{I} denotes the unit tensor and spatial derivatives associated with the reference
 78 configuration are denoted by $\frac{\partial \cdot}{\partial \mathbf{X}} = \nabla_0(\cdot)$. Further, we introduce a scalar-valued
 79 phase-field damage variable $d \in [0, 1]$ such that $d = 0$ for an undamaged, virgin
 80 material and $d = 1$ for a fully damaged, degraded material (See figure 2.1a).
 81 Energy conservation requires the externally supplied energy per time unit \dot{P}^{ext} ,
 82 to be equal to the rate of the internally stored \dot{E} and the dissipated energy \dot{D} :

$$\dot{E} + \dot{D} = \dot{P}^{ext}. \quad (2.1)$$

83 The respective relations for \dot{E} , \dot{D} and \dot{P}^{ext} are defined in the following subsec-
 84 tions. By inserting these relations in equation (2.1), we can derive the governing
 85 equations for our model.

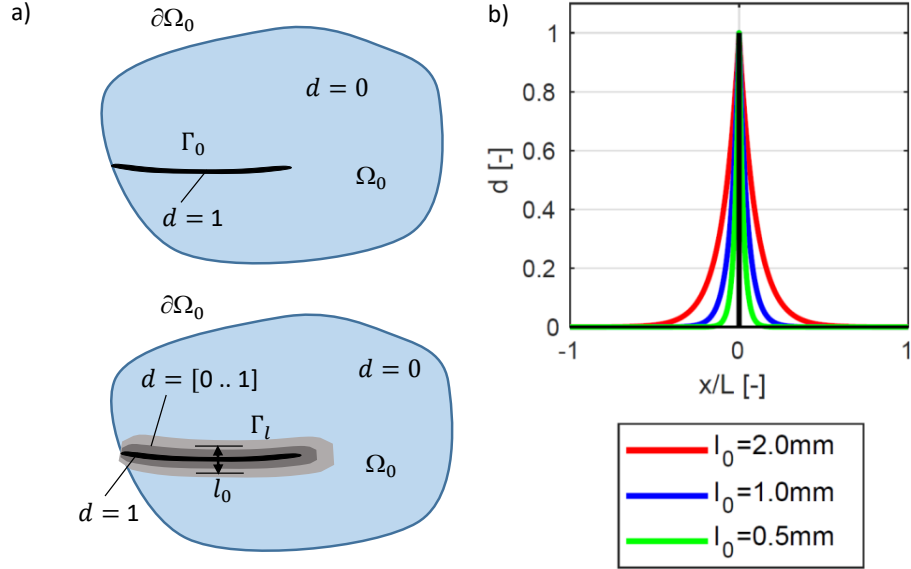


Figure 2.1: a) In a 2-D phase-field damage model a sharp crack Γ_0 is approximated with crack surface Γ_l . b) Damage variable d for a fully developed crack in a 1-D bar with length L . The width of the process zone is controlled by the length scale l_0 .

86 2.1. Rate of internally stored energy

87 The internally stored energy in the bulk reads:

$$E = \int_{\Omega_0 \setminus \Gamma_0} \psi^{bulk} dV = \int_{\Omega_0} g_d \psi^{bulk} dV, \quad (2.2)$$

88 where we have introduced the degradation function g_d^{-1} , which controls the
 89 stiffness of the bulk material as a function of the damage variable d . The
 90 degradation function obeys:

$$\begin{aligned} g_d(d=0) &= 1 \\ g_d(d=1) &= 0 \\ \left. \frac{\partial g_d}{\partial d} \right|_{d=1} &= 0. \end{aligned} \quad (2.3)$$

91 Most studies (e.g. [9], [19],[21], [30]) use the following quadratic degradation
 92 function:

$$g_d = (1 - d)^2. \quad (2.4)$$

93 In contrast, [31] recently introduced the following degradation function:

$$g_d = s((1 - d)^3 - (1 - d)^2) + 3(1 - d)^2 - 2(1 - d)^3, \quad (2.5)$$

94 where $s > 0$ is an additional parameter, which needs to be calibrated. This
 95 degradation function reduces the growth of the damage variable d prior to the
 96 critical stress. For details on the influence of the parameter s , the reader is also
 97 referred to [16]. In chapter 4.4 we investigate the influence of the degradation
 98 function in more detail. As in [31], we set $s = 10^{-4}$.

99 To incorporate rate-dependent effects, we split the strain energy density into an
 100 elastic ψ^{elas} and viscous contribution ψ^{visc} :

$$\psi^{bulk} = \psi^{elas}(\mathbf{F}) + \psi^{visc}(\mathbf{F}, \Phi_\alpha). \quad (2.6)$$

101 Φ_α denotes an internal strain-like tensor, that accounts for the dissipation in
 102 the bulk. It is actually the 3D extension of the 1D strain γ_α in a dashpot of a
 103 Maxwell element (see figure 2.2).

104 Considering the incompressibility of rubbery polymers, the elastic strain energy
 105 density is normally decomposed in an isochoric and volumetric part according

¹Note that we do not add a small constant c to the degradation function ($g_d = g_d + c$), as it is for example employed in [2],[3] or [9], to ensure the stability of the resulting system of equations. The reason is that even a small value for c in combination with a higher order hyperelastic material model has an influence on the solution. Therefore we use an additional Neo-Hookean strain energy potential $\psi^{res} = C_1^{res}(tr(\mathbf{F}^T \cdot \mathbf{F}) - 3)$ in parallel to the elastic ψ^{elas} and viscous ψ^{vis} . This potential does not decline with the degradation function g_d , but the value $C_1^{res} = 5 \cdot 10^{-3} MPa$ is selected so small that it has no influence on the results (not shown here). To present the model as simply as possible we do not include ψ^{res} in the following equations.

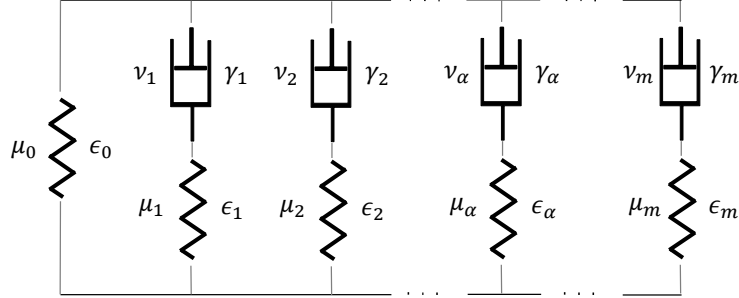


Figure 2.2: Generalized Maxwell-element with m spring-dashpot elements. While μ_α and ϵ_α denote the stiffness and strain in the spring, ν_α and γ_α denote the viscosity and the strain in the dashpot.

106 to:

$$\psi^{elas} = \psi^{iso}(\bar{\mathbf{F}}) + \psi^{vol}(J), \quad (2.7)$$

107 where $\bar{\mathbf{F}} = J^{-1/3}\mathbf{F}$ and $J = \det(\mathbf{F})$. In this contribution, however, we only
 108 consider plane stress cases. With that, the incompressibility constraint can be
 109 applied via substitution in the out of plane deformation [32]. Therefore, we
 110 obtain with $J = 1$ ²:

$$\psi^{elas} = \psi^{iso}(\mathbf{F}). \quad (2.8)$$

111 The rate of the internally stored energy then reads:

$$\dot{E} = \int_{\Omega_0} \left(g_d \frac{\partial \psi^{bulk}}{\partial \mathbf{F}} : \dot{\mathbf{F}} + g_d \sum_{\alpha=1}^m \frac{\partial \psi^{bulk}}{\partial \Phi_\alpha} : \dot{\Phi}_\alpha + \frac{\partial g_d}{\partial d} \psi^{bulk} \dot{d} \right) dV. \quad (2.9)$$

²It is known to the authors, that rubbers do not deform in a perfectly incompressible way ([33], [34]). Recently, researches [35] have shown that decohesion of filler particles from the elastomer matrix is the main reason for the volume growth in tension. With this knowledge, one could postulate a coupling of the compressibility and volume growth to the field of the damage variable. For the sake of simplicity, this is not done in this study and remains an open point for the future.

112 2.1.1. Constitutive equations for the bulk

113 For the elastic part ψ^{elas} we use the strain energy density of a reduced polyno-
114 mial model [36], which reads:

$$\psi^{elas} = \sum_{i=1}^3 C_i (I_1 - 3)^i, \quad (2.10)$$

115 where C_i denote material parameters and $I_1 = tr(\mathbf{F}^T \cdot \mathbf{F})$.

116 Next, we define the viscous contribution $\psi_\alpha^{visc}(\mathbf{F}, \Phi_\alpha)$ with the model for linear
117 viscosity, but finite strains from [37].³ This model is a 3-D generalization of the
118 1-D generalized Maxwell model (figure 2.2) with the viscous stress

$$q_\alpha = \mu_\alpha (\epsilon_0 - \gamma_\alpha) = \nu_\alpha \dot{\gamma}_\alpha. \quad (2.11)$$

119 Since the energy density ψ^{bulk} is split into an elastic and a viscous term, the
120 same applies to the first Piola-Kirchhoff stress:

$$\begin{aligned} \mathbf{P} &= \frac{\partial \psi}{\partial \mathbf{F}} = g_d \frac{\partial \psi^{bulk}}{\partial \mathbf{F}} \\ &= g_d \left(\frac{\partial \psi^{elas}}{\partial \mathbf{F}} + \sum_{\alpha=1}^m \frac{\partial \psi_\alpha^{vis}}{\partial \mathbf{F}} \right) \\ &= g_d \left(\mathbf{P}^\infty + \sum_{\alpha=1}^m \mathbf{Q}_\alpha \right), \end{aligned} \quad (2.12)$$

121 where

$$\mathbf{P}^\infty = 2 \sum_{i=1}^3 i C_i (I_1 - 3)^{(i-1)} \mathbf{F}. \quad (2.13)$$

122 \mathbf{Q}_α denotes the non-equilibrium stress, while \mathbf{P}^∞ denotes the time-infinity
123 stress. From the Clausius-Planck inequality, we extract the rate of dissipation

³There are other, more sophisticated viscoelastic material models for rubber, for example [38], [39] or [40], but these models require an internal Newton-scheme to more accurately account for the viscosity. These models come with substantially larger computational costs. In chapter 4 we show that the model of [37] is accurate enough for our application.

124 of the bulk as:

$$\dot{D}^{visc} = -g_d \sum_{\alpha=1}^m \frac{\partial \psi_{\alpha}^{vis}}{\partial \Phi_{\alpha}} : \dot{\Phi}_{\alpha} = g_d \sum_{\alpha=1}^m \mathbf{Q}_{\alpha} : \dot{\Phi}_{\alpha} \geq 0, \quad (2.14)$$

125 where we have introduced:

$$\mathbf{Q}_{\alpha} = - \frac{\partial \psi_{\alpha}^{vis}}{\partial \Phi_{\alpha}}. \quad (2.15)$$

126 The constraint on the viscous rate of dissipation in equation (2.14) is satisfied
127 by defining the following evolution law for the internal strain-like variables

$$\dot{\Phi}_{\alpha} = \mathcal{V} : \mathbf{Q}_{\alpha}, \quad (2.16)$$

128 where \mathcal{V} denotes a positive definite fourth order tensor containing the inverse
129 viscosity. Incorporating the time derivative of equation (2.11), we formulate the
130 evolution equation within the one-dimensional and linear regime:

$$\dot{q}_{\alpha} + \frac{\mu_{\alpha}}{\nu_{\alpha}} q_{\alpha} = \frac{\mu_{\alpha}}{\nu_{\alpha}} \dot{\epsilon}_0. \quad (2.17)$$

131 Motivated by the solution of the one-dimensional example (equation (2.17)), the
132 viscoelastic stress \mathbf{Q}_{α} is calculated via the following rate equation:⁴

$$\dot{\mathbf{Q}}_{\alpha} + \frac{\mathbf{Q}_{\alpha}}{\tau_{\alpha}} = \dot{\mathbf{P}}_{\alpha}, \quad (2.18)$$

133 where τ_{α} denotes the relaxation time of the α^{th} spring-dashpot element in the
134 3-D generalized Maxwell model. Assuming that the elastic and viscoelastic bulk
135 consist of identical polymer chains, relationship $\psi_{\alpha} = \beta_{\alpha}^{\infty} \psi^{elas}$ is introduced so
136 that:

$$\mathbf{P}_{\alpha} = \frac{\partial \psi_{\alpha}}{\partial \mathbf{F}} = \beta_{\alpha}^{\infty} \frac{\partial \psi^{elas}}{\partial \mathbf{F}} = \beta_{\alpha}^{\infty} \mathbf{P}^{\infty}. \quad (2.19)$$

⁴The viscous response of the material model according to [37] is calculated solely with the rate equation (2.18) and its numerical solution (2.21). Therefore, the model does not require the explicit definition of the fourth order tensor \mathcal{V} and the viscous energy density ψ^{visc} . For the one-dimensional case however, we write $\psi_{\alpha}^{visc} = \frac{1}{2} \mu_{\alpha} (\epsilon_0 - \gamma_{\alpha})^2$ so that $q_{\alpha} = \frac{\partial \psi_{\alpha}^{visc}}{\partial \epsilon_0} = -\frac{\partial \psi_{\alpha}^{visc}}{\partial \gamma_{\alpha}} = \mu_{\alpha} (\epsilon_0 - \gamma_{\alpha})$. This justifies equation (2.15) and comparing equations (2.11) and (2.16), we deduce that \mathcal{V} is the three-dimensional extension of ν_{α}^{-1} .

137 β_α^∞ denote scalar free energy factors. A closed-form solution of rate equation
 138 (2.18) for the time interval $t \in [0, T]$ can be expressed by convolution integrals
 139 as follows:

$$\mathbf{Q}_\alpha = e^{-\frac{T}{\tau_\alpha}} \mathbf{Q}_{\alpha,0} + \int_{t=0}^{t=T} e^{-\frac{T-t}{\tau_\alpha}} \dot{\mathbf{P}}_\alpha dt, \quad (2.20)$$

140 where $\mathbf{Q}_{\alpha,0}$ denotes the instantaneous response. Applying a second-order accu-
 141 rate mid-point rule for the time integration, the viscous stress for the current
 142 time step can then be expressed as:

$$\mathbf{Q}_\alpha = e^{2\zeta_\alpha} \mathbf{Q}_{\alpha,n} + e^{\zeta_\alpha} \beta_\alpha^\infty (\mathbf{P}^\infty - \mathbf{P}_n^\infty), \quad (2.21)$$

143 where $\zeta = \frac{-\Delta t}{2\tau_\alpha}$ and subscripts n denotes converged solutions of the previous
 144 time step t_n .

145 2.2. Rate of Dissipation

146 Additional to the viscous rate of dissipation in the bulk (equation (2.14)), we
 147 introduce dissipation due to crack growth. Following the pioneering work of
 148 [41] and especially [6] for elastomers, we use an energetic approach to fracture.
 149 First, we define G_c as the energy dissipated by the formation of a unit crack
 150 area. Thus, the energy dissipated through crack formation reads:

$$D^{crack} = \int_{\Gamma_0} G_c dA. \quad (2.22)$$

151 Integrating over the fractured surface Γ_0 is difficult and would require constant
 152 remeshing or sophisticated enrichment strategies like XFEM [42]. To circumvent
 153 this surface integral, we approximate the fractured surface $\Gamma_0 \approx \Gamma_l = \int_{\Omega_0} \gamma_l dV$
 154 [10]. With the crack density function $\gamma_l = \gamma_l(d)$, the sharp discontinuity of the
 155 crack is smoothened out to a diffuse topology. The size of this zone is controlled
 156 by the length scale parameter l_0 (See figure 2.1 b). Multiplying G_c with γ_l and
 157 integrating over the domain Ω_0 , the dissipated energy through crack formation
 158 now reads:

$$D^{crack} = \int_{\Gamma_0} G_c dA = \int_{\Omega_0} G_c \gamma_l dV. \quad (2.23)$$

159 As in [9] and [10], we set the crack density function to ⁵:

$$\gamma_l = \frac{1}{2l_0}d^2 + \frac{l_0}{2}(\nabla_0 d \cdot \nabla_0 d), \quad (2.24)$$

160 so that the dissipation rate due to crack formation can be written as:

$$\dot{D}^{crack} = \int_{\Omega_0} G_c \left(\frac{1}{l_0}d\dot{d} + l_0\nabla_0 d \cdot \nabla_0 \dot{d} \right) dV. \quad (2.25)$$

161 Integration by parts and use of the divergence theorem then yields:

$$\dot{D}^{crack} = \int_{\Omega_0} G_c \left(\frac{1}{l_0}d - l_0\nabla_0^2 d \right) \dot{d} dV + \int_{\partial\Omega_0} G_c l_0 \nabla_0 d \cdot \mathbf{n}_0 \dot{d} dA. \quad (2.26)$$

162 \mathbf{n}_0 denotes the outward, unit normal vector and we introduce rate-dependent
163 crack growth dissipation as follows: ⁶

$$\dot{D}^{crack,visc} = \int_{\Omega_0} \kappa_1 \dot{d}^2 dV, \quad (2.27)$$

164 where scalar κ_1 denotes a viscosity parameter.

165 The total rate of dissipation consequently reads:

$$\dot{D} = \dot{D}^{crack} + \dot{D}^{crack,visc} + \dot{D}^{visc} \geq 0. \quad (2.28)$$

166 Enforcing $\dot{d} \geq 0$ (see chapter 3.4) implies that $\dot{D}^{crack} \geq 0$ and hence, that
167 cracks cannot heal. Constraint $\dot{D}^{visc} \geq 0$ was discussed in chapter 2.1.1 and
168 more details can be found in [37]. A positive value of κ_1 finally ensures that
169 $\dot{D}^{crack,visc} \geq 0$.

⁵Alternatively, [43] introduced another crack density function: $\gamma_l = \frac{3}{8l_0}d + \frac{3l_0}{8}(\nabla_0 d \cdot \nabla_0 d)$. This function leads in combination with the quadratic degradation function $g_d = (1-d)^2$ to a reduction of the growth of damage variable d prior to the critical stress [27]. We can reproduce this behavior with the crack density (equation (2.24)) and the degradation function as in equation (2.5). Therefore, we have decided to keep the crack density function constant and only vary the degradation function (see chapter 4.4)

⁶One could introduce higher order viscosity terms with $\kappa_\beta \dot{d}^{2\beta}$ with $\beta > 1$. However, this hardly changes the solution. To keep the model as simple as possible, we set $\beta = 1$.

170 *2.3. Rate of externally applied energy*

171 The rate of externally applied energy reads:

$$\dot{P}^{ext} = \int_{\partial\Omega_0} \mathbf{t}_0 \cdot \dot{\mathbf{u}} \, dA + \int_{\Omega_0} \mathbf{b}_0 \cdot \dot{\mathbf{u}} \, dV, \quad (2.29)$$

172 where \mathbf{t}_0 and \mathbf{b}_0 denote the surface traction and the volumetric body force
173 vector, respectively.

174 *2.4. Balance of mechanical energy*

175 Inserting equations (2.9), (2.28) and (2.29) into equation (2.1) we obtain

$$\begin{aligned} & - \int_{\Omega_0} \left(\nabla_0 \cdot \left(g_d \frac{\partial \psi^{bulk}}{\partial \mathbf{F}} \right) + \mathbf{b}_0 \right) \cdot \dot{\mathbf{u}} \, dV + \int_{\partial\Omega_0} \left(g_d \frac{\partial \psi^{bulk}}{\partial \mathbf{F}} \cdot \mathbf{n}_0 - \mathbf{t}_0 \right) \cdot \dot{\mathbf{u}} \, dA \\ & + \int_{\Omega_0} \left(\frac{\partial g_d}{\partial d} \psi^{bulk} + \frac{G_c}{l_0} d - G_c l_0 \nabla_0^2 d + \kappa_1 \dot{d} \right) \dot{d} \, dV \\ & + \int_{\partial\Omega_0} G_c l_0 \nabla_0 d \cdot \mathbf{n}_0 \dot{d} \, dA + \int_{\Omega_0} \sum_{\alpha=1}^m g_d \left(\mathbf{Q}_\alpha + \frac{\partial \psi^{bulk}}{\partial \Phi_\alpha} \right) : \dot{\Phi}_\alpha \, dV = 0. \end{aligned} \quad (2.30)$$

176 From this, we can extract the macroforce⁷:

$$\nabla_0 \cdot \left(g_d \frac{\partial \psi^{bulk}}{\partial \mathbf{F}} \right) + \mathbf{b}_0 = 0, \quad (2.31)$$

177 and the microforce balance:

$$\frac{\partial g_d}{\partial d} \psi^{bulk} + \frac{G_c}{l_0} d - G_c l_0 \nabla_0^2 d + \kappa_1 \dot{d} = 0. \quad (2.32)$$

178 We want to point out, that equation (2.32) contains a viscous regularization of
179 the damage growth $\kappa_1 \dot{d}$, as well as a rate-dependent driving force:

$$\psi^{bulk} = \psi^{elas} + \psi^{visc} = \int \mathbf{P}^\infty : \dot{\mathbf{F}} \, dt + \sum_{\alpha=1} \int \mathbf{Q}_\alpha : \dot{\mathbf{F}} \, dt. \quad (2.33)$$

180 Finally, the following Neumann boundary conditions may be applied:

$$g_d \frac{\partial \psi^{bulk}}{\partial \mathbf{F}} \cdot \mathbf{n}_0 = \mathbf{t}_0 \quad \text{and} \quad \nabla_0 d \cdot \mathbf{n}_0 = 0. \quad (2.34)$$

181 3. Numerical Implementation

182 3.1. Weak form

183 Next, we transform the macroforce balance (equation (2.31)) and microforce bal-
184 ance (equation (2.32)) to their respective weak form using the standard Galerkin
185 procedure with the test functions $\delta \mathbf{u}$ and δd .

186 We obtain the macroforce balance in the weak form:

$$R_u = \int_{\Omega_0} \mathbf{P} : \nabla_0 \delta \mathbf{u} \, dV - \int_{\Omega_0} \mathbf{b}_0 \cdot \delta \mathbf{u} \, dV - \int_{\Gamma_0} \mathbf{t}_0 \cdot \delta \mathbf{u} \, dS = 0. \quad (3.1)$$

187 Application of the Galerkin procedures to the microforce balance leads to:

$$R_d = \int_{\Omega_0} \left(G_c l_0 \nabla_0^2 d \, \delta d - \frac{\partial g_d}{\partial d} \psi^{bulk} \, \delta d - \frac{G_c}{l_0} d \, \delta d - \kappa_1 \dot{d} \, \delta d \right) dV = 0. \quad (3.2)$$

188 After integration by parts and use of the boundary condition $\nabla_0 d \cdot \mathbf{n}_0 = 0$,
189 equation (3.2) reads:

$$R_d = \int_{\Omega_0} \left(G_c l_0 \nabla_0 d \cdot \nabla_0 \delta d + \frac{\partial g_d}{\partial d} \psi^{bulk} \, \delta d + \frac{G_c}{l_0} d \, \delta d + \kappa_1 \dot{d} \, \delta d \right) dV = 0. \quad (3.3)$$

190 With degradation function $g_d = (1 - d)^2$ and denoting $\eta_1 = \frac{l_0}{G_c} \kappa_1$, we obtain:

$$R_d = \int_{\Omega_0} (l_0^2 \nabla_0 d \cdot \nabla_0 \delta d) \, dV + \int_{\Omega_0} \left(-2(1 - d) \psi^{bulk} \frac{l_0}{G_c} + d + \eta_1 \dot{d} \right) \delta d \, dV = 0. \quad (3.4)$$

⁷By multiplying the stress tensor $\mathbf{P} = g_d \frac{\partial \psi^{bulk}}{\partial \mathbf{F}}$ with the degradation function g_d in equation (2.31), we degenerate the complete bulk response. In case of cyclic loading, this leads to the problem that crack closure is not described correctly. Further, one can see from equation (2.32) that the complete bulk energy is responsible for crack growth, independent of compressive or tensile deformation. As shown in [44], this might lead to an erroneous result in compression. To account for crack surface contacts and to allow crack growth to originate only from tensile deformation, a split of the bulk energy into a positive (tensile) and negative (compression) part is introduced in [44]: $\psi^{bulk} = g_d \psi^+ + \psi^-$. The reader is also referred to [45], in which a spectral decomposition of the strain tensor is used to split the strain energy density into a positive and negative part. Focusing for now on examples that are only exposed to tension, we simplify the model and do not split the energy.

191 Finally, we deduce from equation (3.1) and (3.4) two coupled equations:

$$\begin{aligned} R_u &= R_u(\delta \mathbf{u}, \mathbf{u}, d) = 0 \\ R_d &= R_d(\delta d, \mathbf{u}, d, \dot{d}) = 0, \end{aligned} \quad (3.5)$$

192 which we need to solve.

193 3.2. Linearization

194 To solve equation (3.1) and (3.4), we first discretize the problem in time using
195 a backward Euler scheme:

$$\dot{d} = \frac{d - d_n}{\Delta t}, \quad (3.6)$$

196 resulting in $R_d = R_d(\delta d, \mathbf{u}, d, d_n)$, where d_n denotes the solution of the damage
197 field for previous time step t_n .

198 We then apply a staggered scheme as in [9] and perform an operator split into a
199 mechanical predictor step $ALGO_M$ and damage growth step $ALGO_D$. Accord-
200 ingly, we solve equation (3.1) at time t_{n+1} for the displacement field \mathbf{u} , while
201 keeping the damage field d constant:

$$ALGO_M = \begin{cases} R_u = 0 \\ \dot{d} = 0. \end{cases} \quad (3.7)$$

202 Then, with the updated, but constant displacement field \mathbf{u} , we solve equation
203 (3.4) for the damage field d :

$$ALGO_D = \begin{cases} \dot{\mathbf{u}} = 0 \\ R_d = 0. \end{cases} \quad (3.8)$$

204 Each equation is solved with the Newton-Raphson method. To do so, we need
205 to linearise both equations:

$$\begin{aligned} R_u + \frac{\partial R_u}{\partial \mathbf{u}} \Big|_{\mathbf{u}, d} \cdot \Delta \mathbf{u} &= 0 \\ R_d + \frac{\partial R_d}{\partial d} \Big|_{\mathbf{u}, d} \Delta d &= 0. \end{aligned} \quad (3.9)$$

206 Note that the directional derivatives $\left. \frac{\partial R_u}{\partial d} \right|_{\mathbf{u},d}$ and $\left. \frac{\partial R_d}{\partial \mathbf{u}} \right|_{\mathbf{u},d}$ are neglected. It is
 207 shown in [9] that this scheme is more stable and faster than the monolithic
 208 approach with a full linearization. The only disadvantage is that a sufficiently
 209 small time step is required [46].

210 Calculating the directional derivative for R_u , we obtain:

$$\left. \frac{\partial R_u}{\partial \mathbf{u}} \right|_{\mathbf{u},d} \cdot \Delta \mathbf{u} = \int_{\Omega_0} g_d \left(1 + \sum_{\alpha=1}^m e^{\zeta_\alpha} \beta_\alpha \right) \nabla_0 \delta \mathbf{u} : \mathbf{C}^\infty : \nabla_0 \Delta \mathbf{u} dV, \quad (3.10)$$

211 where

$$\mathbf{C}^\infty = \frac{\partial \mathbf{P}^\infty}{\partial \mathbf{F}} = 2 \sum_{i=1}^3 i C_i (I_1 - 3)^{i-1} \mathcal{I} + 2i(i-1) C_i (I_1 - 3)^{i-2} \mathbf{F} \otimes \mathbf{F}. \quad (3.11)$$

212 Herein, \otimes denotes a dyadic product and $\mathcal{I} = \delta_{ik} \delta_{jl} \mathbf{e}_i \otimes \mathbf{e}_j \otimes \mathbf{e}_k \otimes \mathbf{e}_l$. The
 213 directional derivative for R_d furthermore reads:

$$\begin{aligned} \left. \frac{\partial R_d}{\partial d} \right|_{\mathbf{u},d} \Delta d &= \int_{\Omega_0} l_0^2 \nabla_0 \delta d \cdot \nabla_0 \Delta d dV \\ &+ \int_{\Omega_0} \delta d \left(2\psi^{bulk} \frac{l_0}{G_c} + 1 + \eta_1 \frac{1}{\Delta t} \right) \Delta d dV. \end{aligned} \quad (3.12)$$

214 3.3. Discretization

215 The spatial discretization of the domain is achieved with linear, isoparametric
 216 quadrilaterals. Using N_a to denote the shape function of the a^{th} node and
 217 n_{nodes} to denote the number of nodes, the displacement field \mathbf{u} and the damage
 218 field d are approximated as:

$$\begin{aligned} \mathbf{u} &= \sum_{a=1}^{n_{nodes}} N_a \mathbf{u}_a, \\ d &= \sum_{a=1}^{n_{nodes}} N_a d_a. \end{aligned} \quad (3.13)$$

219 Including these approximations in the weak form, we write:

$$\begin{aligned} R_u(\mathbf{u}, d) + \left. \frac{\partial R_u}{\partial \mathbf{u}} \right|_{\mathbf{u},d} \cdot \Delta \mathbf{u} &= \delta \underline{\mathbf{u}} \cdot (\underline{\mathbf{f}}^u + \underline{\mathbf{K}}^{uu} \cdot \Delta \underline{\mathbf{u}}) = 0 \\ R_d(\mathbf{u}, d) + \left. \frac{\partial R_d}{\partial d} \right|_{\mathbf{u},d} \Delta d &= \delta d \left(\underline{f}^d + \underline{K}^{dd} \Delta d \right) = 0, \end{aligned} \quad (3.14)$$

220 where a single bar under a variable denotes a column and a double bar under a
 221 variable denotes a matrix. Since both equations must hold true for all possible
 222 variations of the displacement field and damage field, we can write the final
 223 nonlinear system of equations as follows:

$$\begin{aligned}\underline{\mathbf{f}}^u + \underline{\mathbf{K}}^{uu} \cdot \Delta \underline{\mathbf{u}} &= \underline{\mathbf{0}} \\ \underline{f}^d + \underline{K}^{dd} \Delta d &= \underline{0},\end{aligned}\quad (3.15)$$

224 where

$$\begin{aligned}\underline{\mathbf{f}}^u &= f_{ia}^u = \int_{\Omega_0} \frac{\partial N_a}{\partial X_j} P_{ij} dV - \int_{\Omega} N_a b_{0,i} dV - \int_{\Gamma} N_a t_{0,i} dS, \\ \underline{f}^d &= f_a^d = \int_{\Omega_0} l_0^2 \frac{\partial N_b d_b}{\partial X_j} \frac{\partial N_a}{\partial X_j} dV + \\ &\quad \int_{\Omega_0} N_a \left(-2(1 - N_b d_b) \psi^{bulk} \frac{l_0}{G_c} + N_b d_b + \eta_1 N_b \frac{d_b - d_{n,b}}{\Delta t} \right) dV, \\ \underline{\mathbf{K}}^{uu} &= [K_{ab}^{uu}]_{ik} = \int_{\Omega_0} g_d \left(1 + \sum_{\alpha=1}^m e^{\zeta_\alpha} \beta_\alpha \right) \frac{\partial N_a}{\partial X_j} C_{ijkl}^\infty \frac{\partial N_b}{\partial X_i} dV, \\ \underline{K}^{dd} &= [K_{ab}^{dd}] = \int_{\Omega_0} l_0^2 \frac{\partial N_a}{\partial X_i} \frac{\partial N_b}{\partial X_i} dV \\ &\quad + \int_{\Omega_0} N_a \left(2\psi^{bulk} \frac{l_0}{G_c} + 1 + \frac{\eta_1}{\Delta t} \right) N_b dV.\end{aligned}\quad (3.16)$$

225 3.4. Enforcing Irreversibility with an Active Set Method

226 It is essential that damaged material is prevented from healing. We examine
 227 this problem for a 1D rate-dependent formulation without gradient terms [23].
 228 Equation (2.32) then reads with $g_d = (1 - d)^2$:

$$2(1 - d)\psi^{bulk} - \frac{G_c}{l_0} d - \kappa_1 \dot{d} = 0. \quad (3.17)$$

229 Furthermore, we assume a linear-elastic material $\psi^{bulk} = \frac{1}{2} E \epsilon^2$, where E and ϵ
 230 denote the Young's modulus and the strain, respectively. Inserting $\dot{d} = \frac{d - d_n}{\Delta t}$,
 231 we obtain the following damage variable at the end of the current time step:

$$d = \frac{2\psi^{bulk} + \frac{\kappa_1 d_n}{\Delta t}}{2\psi^{bulk} + \frac{G_c}{l_0} + \frac{\kappa_1}{\Delta t}}. \quad (3.18)$$

232 If we now apply strains as depicted in figure 3.1a), we would observe healing
 233 for a decrease of the strain, until the material is entirely healed at $\epsilon_{(t=4s)} = 0$.
 234 To avoid healing (i.e. to guarantee irreversibility of the damage), [9] introduced
 235 the following history variable:

$$H = \max_{s=[0,t]} [\psi^{bulk}(s)], \quad (3.19)$$

236 such that equation (3.18) reads:

$$d = \frac{2H + \frac{\kappa_1 d_n}{\Delta t}}{2H + \frac{G_c}{l_0} + \frac{\kappa_1}{\Delta t}}. \quad (3.20)$$

237 The results are presented in figure 3.1b) - d). Even though history variable H
 238 remains constant for a decrease of the strain, damage variable d continues to
 239 grow. Therefore, instead of using the history variable H , we use an active set
 240 method [47] to enforce $\dot{d} \geq 0$ as a constraint (as in [48]).

241 The part of the system of equations associated with the computation of the
 242 damage variable is partitioned into a set $\mathcal{A} = \{i | d < d_n\}$ with active constraints
 243 and with complementary inactive constraint \mathcal{I} . Within each Newton iteration,
 244 we solve the reduced system of the inactive constraint as follows:

$$\Delta d_{\mathcal{I}} = - \left(\underline{K}^{dd} \right)_{\mathcal{I}\mathcal{I}}^{-1} f_{\mathcal{I}}^d, \quad (3.21)$$

245 while setting $\Delta d_{\mathcal{A}} = 0$. The active set \mathcal{A} is updated within each iteration until
 246 the constraint is fulfilled at every node. The procedure is detailed in the pseudo-
 247 code presented in algorithm 1. As can be seen in figure 3.1, the damage stops
 248 growing for a decreasing strain using this method.

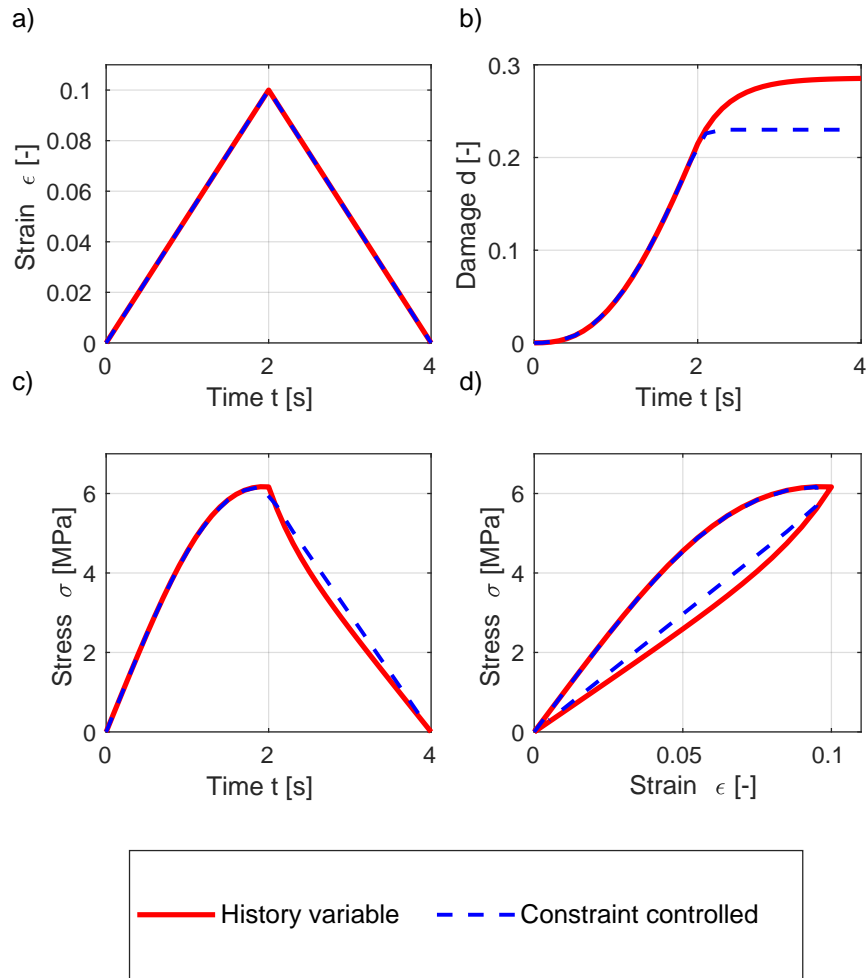


Figure 3.1: Solutions of a rate-dependent 1-D phase-field model calculated with a history field and a constraint on \dot{d} a) Applied strain over time b) Evolution of the damage variable over time c) Evolution of the stress over time d) Strain-stress response of the system.

Algorithm 1 Active Set Newton Method

```

1:  $\mathcal{A} = \emptyset$ ,  $\mathcal{I} = all$ 
2: while  $Res > tol$  do
3:   Assemble  $\underline{f}^d, \underline{K}^{dd} \leftarrow \mathbf{u}, \underline{d}^{k-1}, \underline{d}_n$ 
4:    $Res = \|\mathbf{f}_{\mathcal{I}}^d\|$ 
5:    $\Delta \underline{d}_{\mathcal{I}}^k = - \left( \underline{K}^{dd} \right)_{\mathcal{I}\mathcal{I}}^{-1} \underline{f}_{\mathcal{I}}^d$ 
6:   while  $\Delta \underline{d}_{\mathcal{I}} < 0$  do
7:      $\mathcal{A} = \mathcal{A} \cup (\Delta \underline{d}_{\mathcal{I}} < 0)$ 
8:      $\Delta \underline{d}_{\mathcal{A}} = 0$ 
9:      $\mathcal{I} = \mathcal{I} \cap \mathcal{A}$ 
10:     $\Delta \underline{d}_{\mathcal{I}}^k = - \left( \underline{K}^{dd} \right)_{\mathcal{I}\mathcal{I}}^{-1} \underline{f}_{\mathcal{I}}^d$ 
11:    $\underline{d}^k = \underline{d}^{k-1} \mp \Delta \underline{d}^k$ 
12:    $k = k + 1$ 
13: Return

```

249 *3.5. Corrector scheme and time adaptivity*

250 As detailed in chapter 3.2, equations (3.1) and (3.4) are solved in the staggered
 251 scheme as proposed by [9]. A disadvantage of this approach is a need of a suffi-
 252 ciently small time step. To decrease the calculation time we introduce adaptive
 253 time stepping. This reduces or increases the time step depending on the growth
 254 of the damage variable from one converged time step to another.

255 Furthermore, we have incorporated a corrector scheme according to [28] with it-
 256 erations between the macro- and microforce balance within one time step. This
 257 means that for each time step we solve first for the displacements \mathbf{u}^j and then
 258 for the damage field \underline{d}^j , but instead of proceeding to the next time step, we
 259 calculate again the displacement field \mathbf{u}^{j+1} with the updated damage field \underline{d}^j .
 260 Next, we update damage field \underline{d}^{j+1} . The scheme only proceeds to the next time
 261 step if the change of the displacement field and damage field from one iteration
 262 to the other is smaller than a predefined tolerance. The pseudo code for this
 263 scheme is presented in algorithm 2, where Δt_{min} and Δt_{max} denote the limits
 264 for the time step size. Note that $ALGO_D$ is solved with the active-set method
 265 as outlined in algorithm 1.

266 To highlight the advantage of the scheme, we compare in figure 3.2 the global
 267 force response of a single-edge notched tensile test for different time steps. De-
 268 tails of the geometry can be found in figure 4.1a), while the fracture parameters
 269 are set to $l_0 = 1.50mm$, $G_c = 3.0N/mm$ and $\eta_1 = \frac{l_0}{G_c} \kappa_1 = 0.05$. Further, we
 270 set $\Delta t_{max} = 50\Delta t_{min}$ and $tol_3 = 5.0 \cdot 10^{-4}$. The staggered scheme converges
 271 for a sufficiently small time step Δt_{min} to a stable solution. By applying the
 272 corrector scheme, we can use a substantially larger time step. In combination
 273 with the time step adaptivity, the calculation time is approximately reduced by
 274 a factor of 6. Note that we did not use the time step adaptivity for the first
 275 three results, but did the simulation with $\Delta t = \Delta t_{max}$ until $\lambda = 1.05$ and then
 276 $\Delta t = \Delta t_{min}$. Therefore, the time step adaptivity not only reduces the compu-
 277 tation time but also the required user input, since the timing of the time step
 278 change depends on each case.

Algorithm 2 Corrector scheme + time step adaptivity

```

1:  $\underline{\mathbf{u}}^0 = \underline{\mathbf{u}}^1 = \underline{\mathbf{u}}^2 = \underline{\mathbf{u}}_n$ 
2:  $\underline{d}^0 = \underline{d}^1 = \underline{d}^2 = \underline{d}_n$ 
3:  $j = 2$ 
4: while ( $Res_1 > tol_1$  or  $Res_2 > tol_2$ ) do
5:   solve  $ALGO_M(\underline{\mathbf{u}}^{j-1}, \underline{d}^{j-1})$  for  $\underline{\mathbf{u}}^j$ 
6:   solve  $ALGO_D(\underline{\mathbf{u}}^j, \underline{d}^{j-1})$  for  $\underline{d}^j$ 
7:    $Res_1 = \frac{abs(\|\underline{d}^{j-1} - \underline{d}^{j-2}\| - \|\underline{d}^j - \underline{d}^{j-1}\|)}{1 + \|\underline{d}^{j-1} - \underline{d}^{j-2}\|}$ 
8:    $Res_2 = \frac{abs(\|\underline{\mathbf{u}}^{j-1} - \underline{\mathbf{u}}^{j-2}\| - \|\underline{\mathbf{u}}^j - \underline{\mathbf{u}}^{j-1}\|)}{1 + \|\underline{\mathbf{u}}^{j-1} - \underline{\mathbf{u}}^{j-2}\|}$ 
9:    $j = j + 1$ 
10: if  $max(\underline{d}^j - \underline{d}_n) > tol_3$  then
11:    $\Delta t = max(\Delta t_n / 2; \Delta t_{min})$ 
12: else
13:    $\Delta t = min(1.2 \Delta t_n; \Delta t_{max})$ 
14:  $t = t_n + \Delta t$ 
15:  $\underline{\mathbf{u}} = \underline{\mathbf{u}}^j$ 
16:  $\underline{d} = \underline{d}^j$ 
17: Return

```

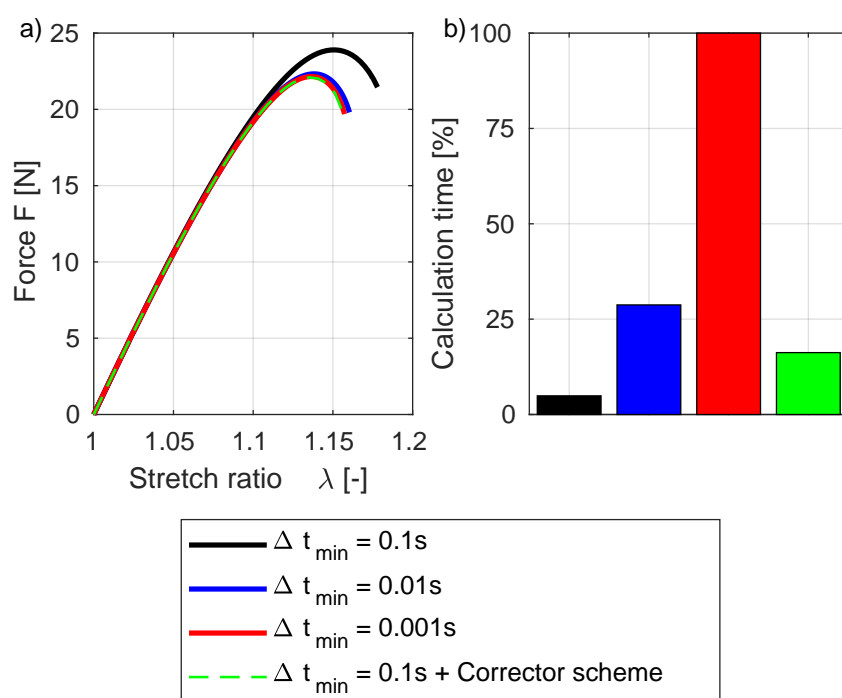


Figure 3.2: a) Force to stretch-ratio response for various time steps for a single notch tensile test. The solution converges for a sufficiently small time step. b) Normalized calculation time for various time steps.

279 4. Results

280 We start this chapter by introducing the experiments used to identify the ma-
281 terial parameters. Next, we discuss the influence of the degradation function.
282 Length scale l_0 as a material parameter is set to a finite size and we argue that
283 the value obtained from our experiments fits the material microstructure. Last,
284 we validate the model with the identified parameter set by performing additional
285 numerical tests and compare the results to their experimental counterparts.

286 4.1. Experiments

287 An ethylene propylene diene monomer rubber (EPDM) is tested in all experi-
288 ments, which are of a plane stress nature. The strains are measured using a laser
289 extensometer and we also use digital image correlation (DIC) to measure strain
290 fields. Because crack growth in EPDM is highly dependent on temperature, we
291 have conducted all experiments at a constant temperature of $20^\circ C$.

292 First, the bulk parameters are identified using uniaxial tensile tests, which are
293 performed according to ISO 37:2002 with dumbbell specimens. The sample
294 thickness is $2mm$, while the length is $20mm$ and the width is $4mm$. The results
295 for three clamp velocities are presented in figure 4.2. We indeed observe that
296 an increase in the loading rate yields larger stresses.

297 Single (SENT) and double-edge notch tensile tests (DENT) are performed to
298 identify the phase-field parameters (see figure 4.1). Additionally to the force-
299 displacement response, we can measure local strains near the crack tip using
300 DIC. Due to the specifications of the camera, we only measure local strain fields
301 during the SENT test for a clamp velocity of $25mm/min$. Local strain fields
302 are computed using GOM Correlate software [49].

303 4.2. Identification of bulk material parameters

304 First, we calibrate the bulk material parameters. We use the hyperelastic model
305 of [36] with three parameters according to equation (2.10) in combination with
306 two Maxwell-elements [37]. The bulk parameters are identified using the least-
307 squares method for which the minimization is performed using the Nelder-Mead

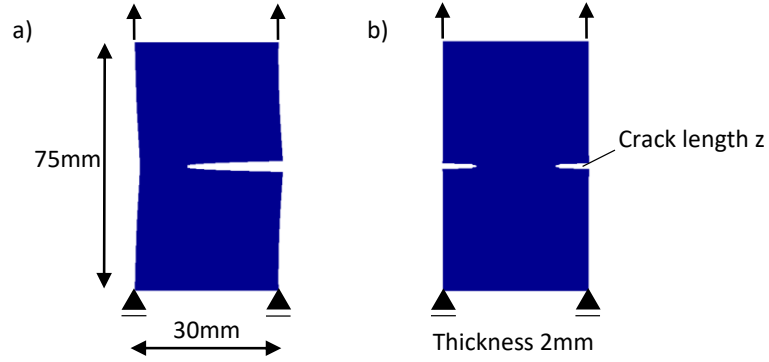


Figure 4.1: a) Single-edge notch tensile (SENT) test with crack length $z = 20\text{mm}$ b) Double-edge notch tensile (DENT) test with variable crack length z .

308 simplex approach in MATLAB [50]. Table 4.1 and figure 4.2 display the bulk
 309 parameters and the associated material responses, respectively.

Table 4.1: Identified material parameters for visco-hyperelastic model.

C_1 [MPa]	C_2 [MPa]	C_2 [MPa]
0.9600	0.0430	6.316×10^{-06}

β_1 [-]	β_2 [-]	τ_1 [s]	τ_2 [s]
0.40642	0.0284	4.9776	449.3075

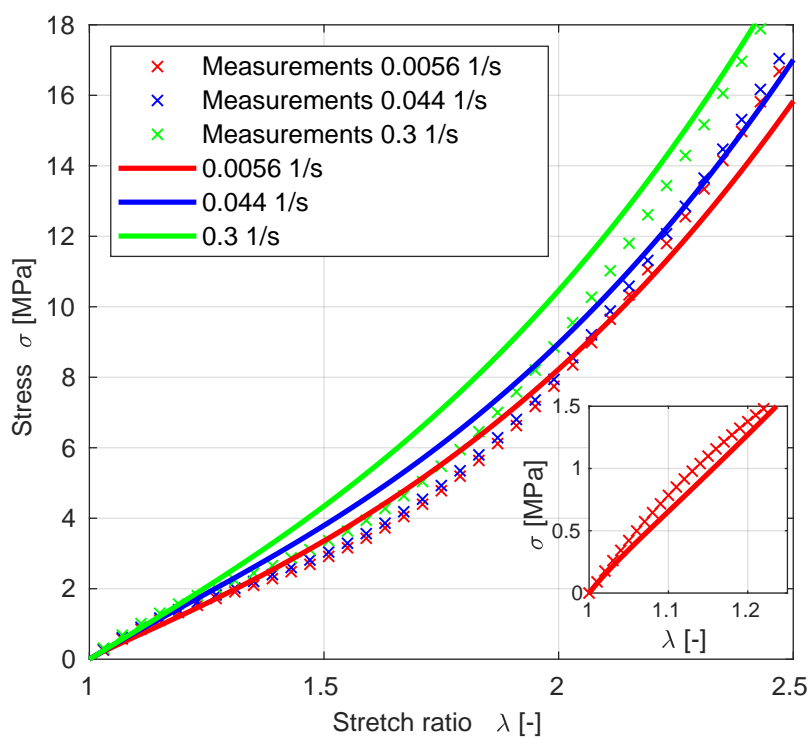


Figure 4.2: Uniaxial tensile test results: Averaged experimental results and material model responses for 3 clamp velocities after the bulk parameters are identified. All tests are performed at 20°C and at least 5 samples were tested per clamp velocity. The zoom in the right-bottom corner shows the result for a clamp velocity of 0.0056s^{-1} .

310 4.3. Identification of phase-field damage parameters

311 The fracture parameters are identified using the force-displacement response of
 312 the SENT tests (figure 4.1) with an initial crack length of $20mm$ and the DENT
 313 test with an initial crack of $7mm$. Measurements are recorded during the SENT
 314 tests with clamp velocities of $25mm/min$ (test data $y_{k,25mes}$) and $200mm/min$
 315 ($y_{k,200mes}$) and during the DENT test with a clamp velocity of $75mm/min$
 316 ($y_{k,75mes}$). Lastly, we include the local strains in front of the crack tip measured
 317 via DIC ($y_{k,25mesDIC}$). The local strains are measured at a global displacement
 318 of $10.5mm$, which is the point of crack nucleation in our experiments. By
 319 application of the least squares method, we define the residual to be minimized:

320

$$\begin{aligned}
 RES = & w \left(\sum_{k=1}^{n_{mes25}} \left(\frac{y_{k,25mes} - y_{k,25}}{y_{k,25mes}} \right)^2 + \sum_{k=1}^{n_{mes200}} \left(\frac{y_{k,200mes} - y_{k,200}}{y_{k,200mes}} \right)^2 \right. \\
 & + \sum_{k=1}^{n_{mes75}} \left(\frac{y_{k,75mes} - y_{k,75}}{y_{k,75mes}} \right)^2 \left. \right) \\
 & + (1 - w) \sum_{k=1}^{n_{mes25DIC}} \left(\frac{y_{k,25mesDIC} - y_{k,25DIC}}{y_{k,25mesDIC}} \right)^2,
 \end{aligned} \tag{4.1}$$

321 where subscript *mes* denotes experimentally measured values. The scalar $w =$
 322 0.25 is introduced to weigh the force-displacement results with respect to the
 323 measured local strains. The minimization of the residual is performed using a
 324 genetic algorithm [51] and the identified fracture parameters are presented in
 325 table 4.2.

Table 4.2: Identified fracture parameters for degradation function $g_{d,1} = (1 - d)^2$.

G_c [N/mm]	l_0 [mm]	η_1 [-]
7.819	0.55040	0.10610

326 *4.3.1. Single-edge notch tensile tests*

327 In this subsection, we show the experimental and numerical results of the cases
 328 used to calibrate the fracture parameters. The results for the DENT test with
 329 an initial crack $z = 7mm$ and a clamp velocity of $75mm/min$ ($y_{k,75mes}$) are
 330 presented in section 4.7.1 (figure 4.14b).

331 The evolution of the phase-field parameter d and therefore the crack propagation
 332 for the SENT test is shown in figure 4.3. By omitting elements with an average
 333 damage variable $d_{average} > 0.95$, the crack is made visible.

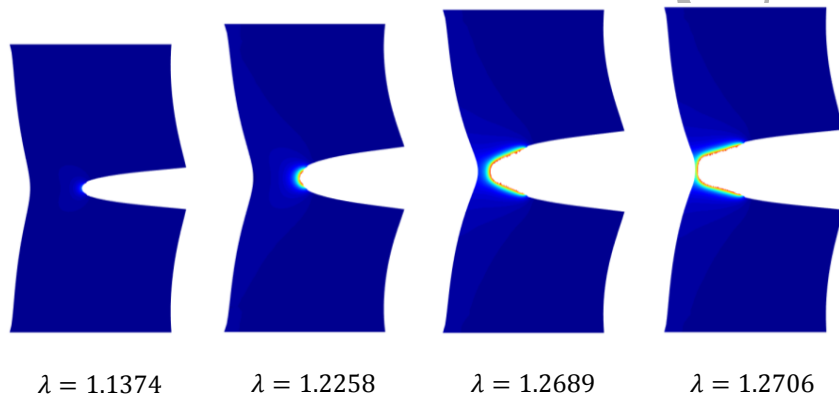


Figure 4.3: SENT test with initial crack size $z = 20mm$ and clamp velocity $25mm/min$:
 Numerically predicted crack growth over stretch ratio λ .

334 Next, we compare the force to stretch ratio response for a clamp velocity of
 335 $25mm/min$ and $200mm/min$. As can be seen in figure 4.4, the model is clearly
 336 capable of tracking the increase of the maximum tearing force for an increase

337 of the clamp velocity.⁸

338 The experimental results for both velocities are compared to the numerical
339 results in figure 4.5 and we observe a sufficiently accurate agreement.

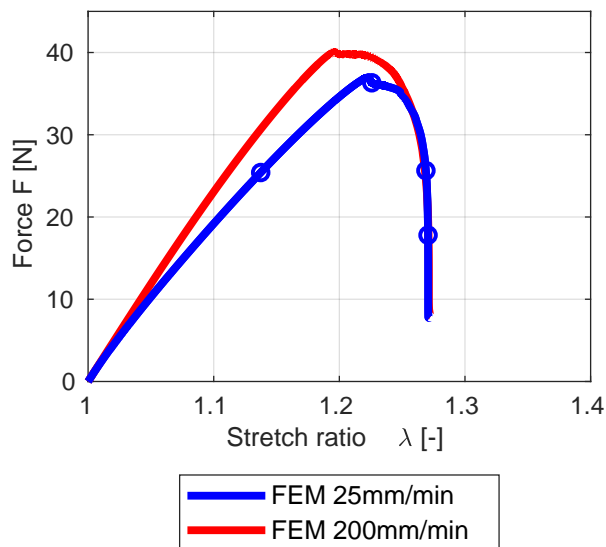


Figure 4.4: SENT test: Tearing force to stretch ratio for loading rates $25\text{mm}/\text{min}$ and $200\text{mm}/\text{min}$. The circles highlight the specific points at which we plot the damage variable in figure 4.3.

340 By application of the DIC technology, we were able to measure local strains
341 near the crack tip. In figure 4.6 we compare the strains in the y-direction at a

⁸We see for both velocities a small kink in the force response after reaching the maximum tearing force. This kink is not observed in our experiments (figure 4.5). By tweaking our material parameters we could design a smooth force response. In detail, we could either increase the length-scale l_0 , reduce the fracture toughness G_c , use a Neo-Hookean hyperelastic material model, or increase or decrease rate-dependency $(\eta_1, \tau_\alpha, \beta_\alpha)$. To the best of the authors' knowledge, this problem has not been reported before. [20] and [52] have investigated a SENT specimen with a higher-order hyperelastic material model [53] and rate-independent phase-field damage model. However, this kink was not occurring close to the maximum tearing force, but at a later stage, and they did not comment on it. We decide not to tweak the material parameters and postpone an investigation to a later work.

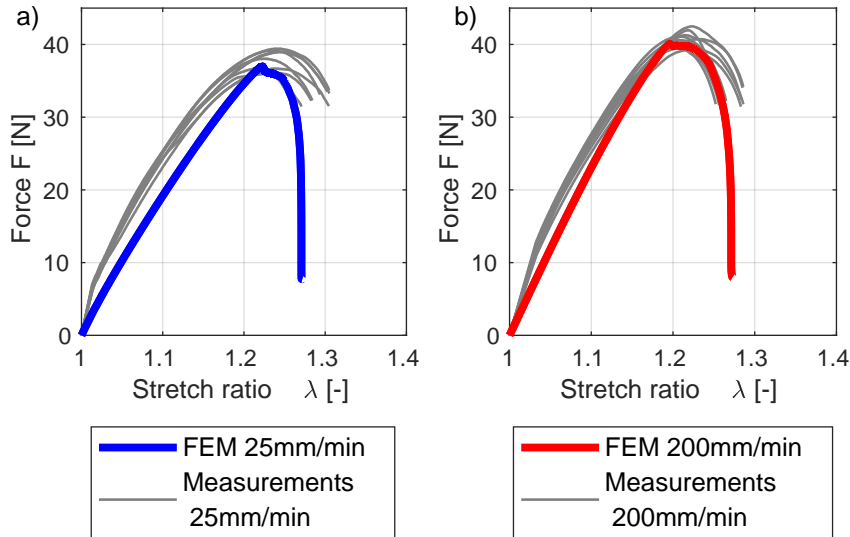


Figure 4.5: SENT test: Numerical and experimental results a) Force to stretch-ratio response for $25\text{mm}/\text{min}$. b) Force to stretch-ratio response for $200\text{mm}/\text{min}$.

342 global displacement of 10.5mm . To better quantify the results we look at two
 343 paths, denoted by x and y in figure 4.6. The experimental and numerical results
 344 of these paths are plotted in figure 4.7. Comparing the size of the process zone,
 345 id est the region of large deformation, the accurate fit supports our identified
 346 value of the length scale parameter.

347 4.4. Influence of degradation function g_d

348 As shown in [54] and [55], the degradation function impacts the results and may
 349 be tailored to fit experimental measurements. Therefore, we also investigate
 350 the responses for the following two degradation functions: $g_{d,2} = (1 - d)^3$ and
 351 $g_{d,3} = m[(1 - d)^3 - (1 - d)^2] + 3(1 - d)^2 - 2(1 - d)^3$. These degradation functions,
 352 together with the quadratic degradation function $g_{d,1} = (1 - d)^2$, are presented
 353 in figure 4.8a). The cubic degradation function makes the damage grow faster
 354 than the original quadratic degradation function and the combined one makes
 355 the damage grow slower. Figure 4.8b) shows the material responses of a 1D

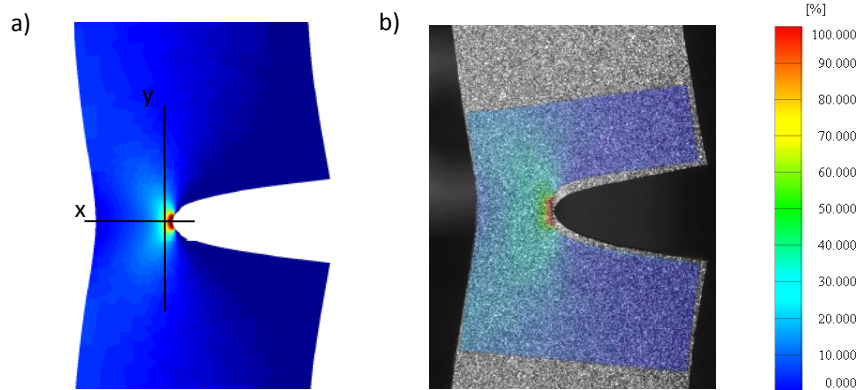


Figure 4.6: a) Numerically predicted Green-Strain E_{yy} . The lines x and y indicate the path along which we plot in figure 4.7 the strains. Additionally, the y -direction is corresponding with the y -path. b) Green-Strain E_{yy} measured via DIC.

bar with Young's modulus $E = 2000MPa$, length-scale $l_0 = 0.1mm$, viscosity $\kappa_1 = 0$ and fracture toughness $G_c = 2N/mm$. The combined quadratic, cubic degradation function clearly results in a nearly linear-elastic behavior before the onset of failure. The failure response itself is substantially more brittle than for the two other degradation functions. The responses for the other two degradation functions show that the onset of failure does not correspond to the maximum stress. The cubic degradation function furthermore results in a more ductile response than the quadratic one.

The identified fracture parameters for the degradation functions $g_{d,2}$ and $g_{d,3}$ are summarized in tables 4.3 and 4.4, respectively. Since the residual with the quadratic degradation function is the smallest, which is graphically illustrated for the SENT test at $25mm/min$ in figure 4.9, we select the quadratic one. For the interested reader, we have included the results for degradation functions $g_{d,2}$ and $g_{d,3}$ for the SENT test at $200mm/min$ and for the DENT tests in Appendix A.

Interestingly, we observe that the length scale parameter is of the same order of magnitude for all degradation functions ($l_0 \approx 0.55mm$). This strengthens the theory that the length scale parameter is indeed a parameter that depends on

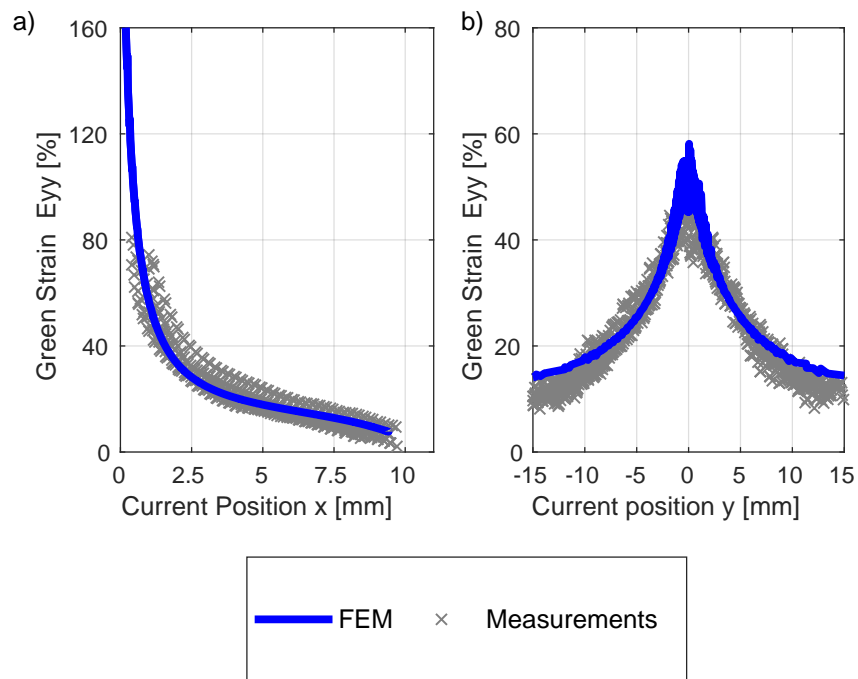


Figure 4.7: Comparison of experimental and numerical local Green-Strain E_{yy} : a) Local strains for path x b) Local strains for path y.

374 the microstructure of the material.

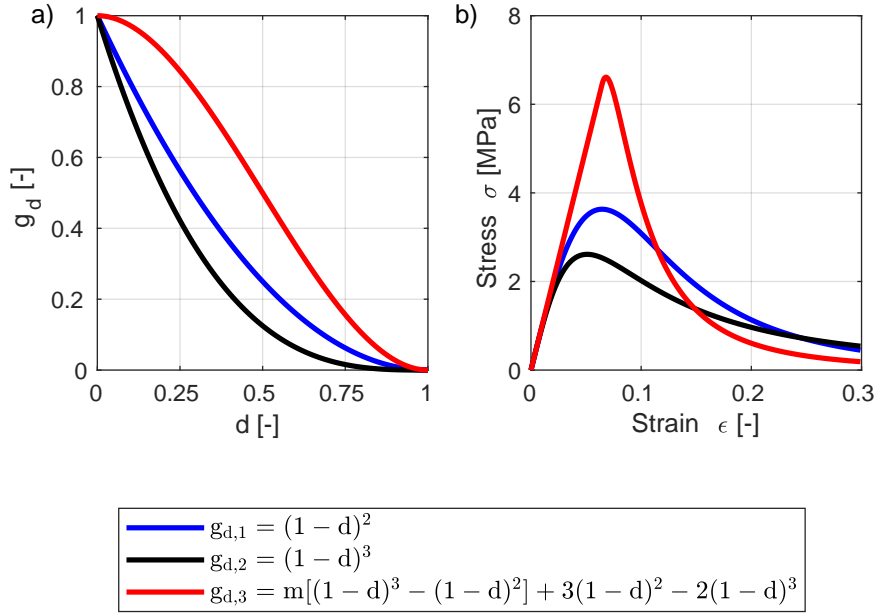


Figure 4.8: a) Evolution of the degradation functions for damage variable $d \in [0, 1]$ b) Stress-strain response for a linear elastic 1D case.

Table 4.3: Identified fracture parameters for degradation function $g_{d,2} = (1 - d)^3$.

G_c [N/mm]	l_0 [mm]	η_1 [-]
12.31	0.564	0.0766

Table 4.4: Identified fracture parameters for degradation function $g_{d,3} = m[(1 - d)^3 - (1 - d)^2] + 3(1 - d)^2 - 2(1 - d)^3$.

G_c [N/mm]	l_0 [mm]	η_1 [-]
4.426	0.693	0.0651

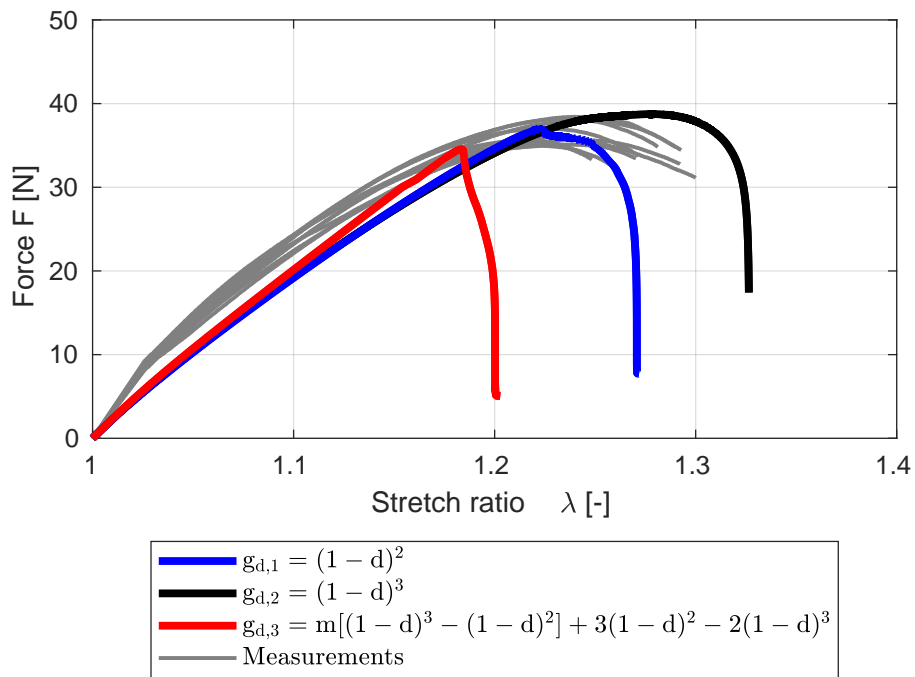


Figure 4.9: SENT responses ($25\text{mm}/\text{min}$) for the three degradation functions together with the experimental data.

375 4.5. Interpretation of length scale l_0

376 Optimization with genetic algorithm leads to a length scale parameter $l_0 =$
 377 $0.55mm$. First, to validate that the length scale is not a solution of the grid
 378 size, we study in figure 4.10 the influence of the spatial discretization on the
 379 results. We see that the maximum force, as well as the local strains, do not
 380 depend on the mesh.

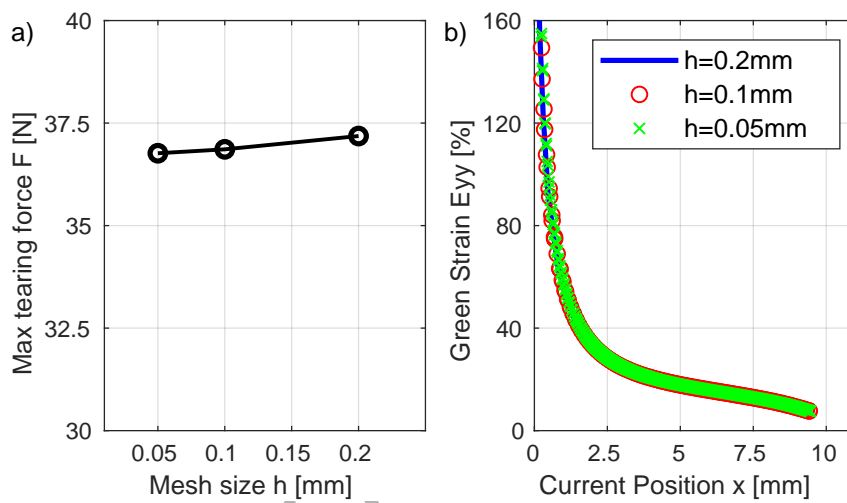


Figure 4.10: SENT test $25mm/min$: Comparison of three different mesh sizes h : a) Maximum tearing force b) Local strains E_{yy} for path x .

381 The EPDM rubber we use is reinforced with carbon particles. Consequently,
 382 we consider the microstructure of the rubber as a composite made of a polymer
 383 matrix and rigid carbon particles. Applying a high strain to this composite,
 384 debonding of the polymer matrix from the particles was observed [35]. We
 385 assume that the high local strains in front of the crack tip lead to debonding in
 386 a finite region. The resulting microcracks are then quantified in our model with
 387 a growth of the damage variable d (figure 4.11).

388 The average diameter of carbon particle agglomerates in our rubber is $15\mu m$,
 389 but they can be as large as $50\mu m$. Estimating that debonding occurs at several
 390 agglomerates in parallel, we argue that a length scale parameter of $0.55mm$ is

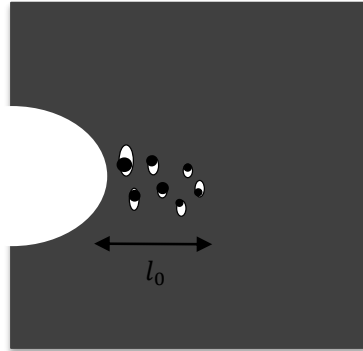


Figure 4.11: Due to many small inclusions, we expect debonding and subsequently microcracks to occur at a finite distance to the crack tip.

391 reasonable.

392 For comparison, [56] has predicted the characteristic length of damage localiza-
 393 tion in a natural rubber to be 0.2mm . Measuring the heat source at the crack tip
 394 for a filled styrene-butadiene rubber (SBR) under cyclic loading a localization
 395 in a zone of $0.44 \times 0.32\text{mm}^2$ is found in [57]. By assuming that the heat is gen-
 396 erated by the formation of microcracks our argument of a finite process zone is
 397 strengthened. Tests on a filled SBR are also performed in [58]. Nanocavitation
 398 was observed by an X-ray microbeam scan and was found to be at least $300\mu\text{m}$
 399 in front of the crack tip. Due to the fact that all these references have presented
 400 measured length scales of a similar magnitude for rubbers, the identified value
 401 can indeed be considered as appropriate.

402

403 4.6. Balance of mechanical energy and dissipation during crack growth

404 To verify the thermodynamical consistency of the model, id est the fulfillment of
 405 the first (2.1) and second law of thermodynamics (2.28), the energetic signature
 406 of the SENT test is shown in figure 4.12 for a clamp velocity of $25\text{mm}/\text{min}$
 407 and in figure 4.13 for $200\text{mm}/\text{min}$. The maximum relative error of the ex-
 408 ternal power and the sum of the internally stored and dissipated energy is

409 $err_{25} = \max\left(\frac{\sum(E+D)-P^{ext}}{P^{ext}}\right) = 0.0052$ for $25\text{mm}/\text{min}$ and $err_{200} = 0.0022$
 410 for $200\text{mm}/\text{min}$. Since the error is small, we consider the balance of mechani-
 411 cal energy fulfilled.
 412 Note that we are only able to calculate the sum:

$$D^{visc} + E^{visc} = g_d \sum_{\alpha=1} \int \mathbf{Q}_{\alpha} : \dot{\mathbf{F}} dt, \quad (4.2)$$

413 because the material model of [37] does not include a split of the deformation
 414 tensor into an elastic and inelastic part. For clamp velocity $200\text{mm}/\text{min}$, a
 415 slight decrease of the sum $D^{visc} + E^{visc}$ is observed after reaching the maximum
 416 tearing force. D^{visc} can be seen as the dissipated energy in the dashpot (see
 417 figure 2.2), while E^{visc} is the stored energy in the spring in series to the dashpot.
 418 Especially for fast loading, such as $200\text{mm}/\text{min}$, the dashpot has no time to
 419 relax so that the energy stored in the spring is high. Therefore, we conclude that
 420 sum $D^{visc} + E^{visc}$ decreases due to the degradation of E^{visc} while $\dot{D}^{visc} \geq 0$.
 421 Since dissipation due to crack growth (D^{crack} and $D^{visc,crack}$) is increasing for
 422 the entire loading process, the second law of thermodynamics is fulfilled.
 423 Further, we want to highlight that most of the viscous dissipation is caused
 424 by the bulk D^{visc} , and not by viscous crack resistance $D^{visc,crack}$. This shows
 425 the necessity to not only include the time and rate-dependency in the damage
 426 evolution, but also in the constitutive bulk response.

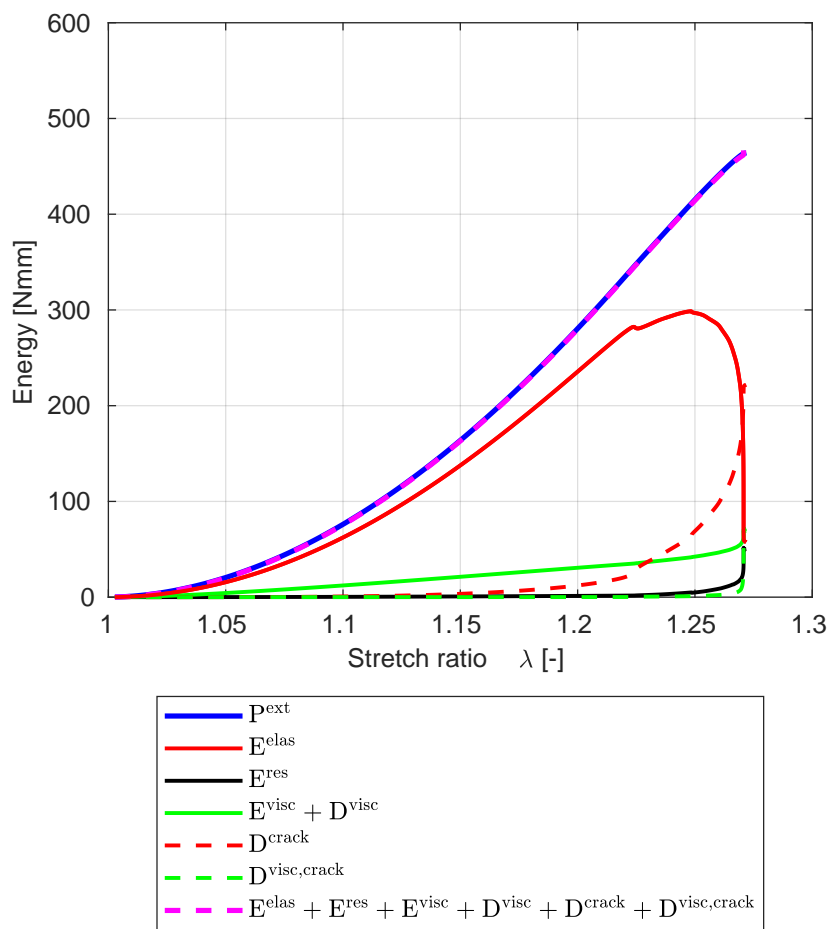


Figure 4.12: SENT test 25mm/min: Energy of the system

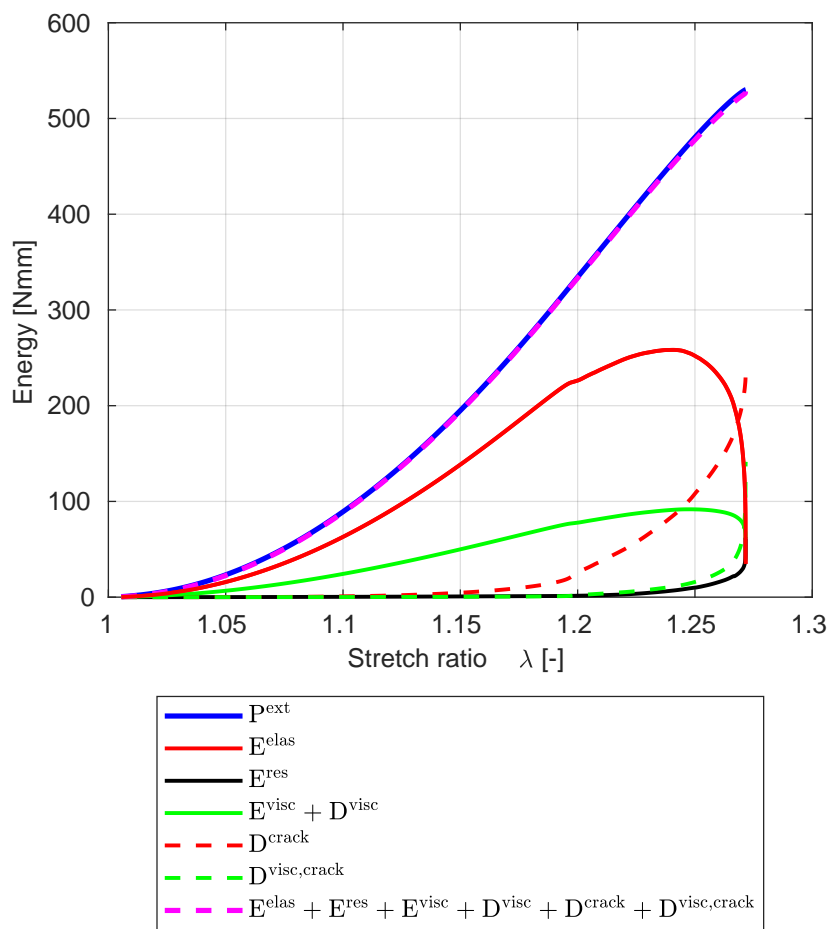


Figure 4.13: SENT test 200mm/min: Energy of the system

427 *4.7. Validation*

428 In this subsection, we use the optimized set of material parameters (table 4.2)
429 and perform additional validation tests with different geometries and loading
430 rates.

431 *4.7.1. Double-edge notch tensile test with variable initial crack length*

432 First, we investigate the DENT test, as depicted in figure 4.1, and change the ini-
433 tial crack size. This experiment was first reported by [59] and later used by oth-
434 ers ([1], [19],[20]) to validate damage models for rubber. We have repeated these
435 experiments for our EPDM rubber with a crack size $z = [3mm; 5mm; 7mm; 9mm]$
436 and a clamp velocity of $75mm/min$. In figure 4.14 we present the measured and
437 calculated force to stretch ratio response for all crack sizes. We observe a good
438 agreement between the experimental data and computed predictions. Although
439 the maximum stretch is slightly underestimated for all initial crack lengths, the
440 maximum tearing force is accurately predicted.

441 *4.7.2. Double-edge notch tensile test with variable loading rate*

442 Next, we continue with an initial crack length of $z = 7mm$ and change the load-
443 ing rate $[25 .. 200mm/min]$ for the DENT test (figure 4.15). The experimentally
444 observed increase of the maximum tearing force is successfully captured by the
445 proposed model.

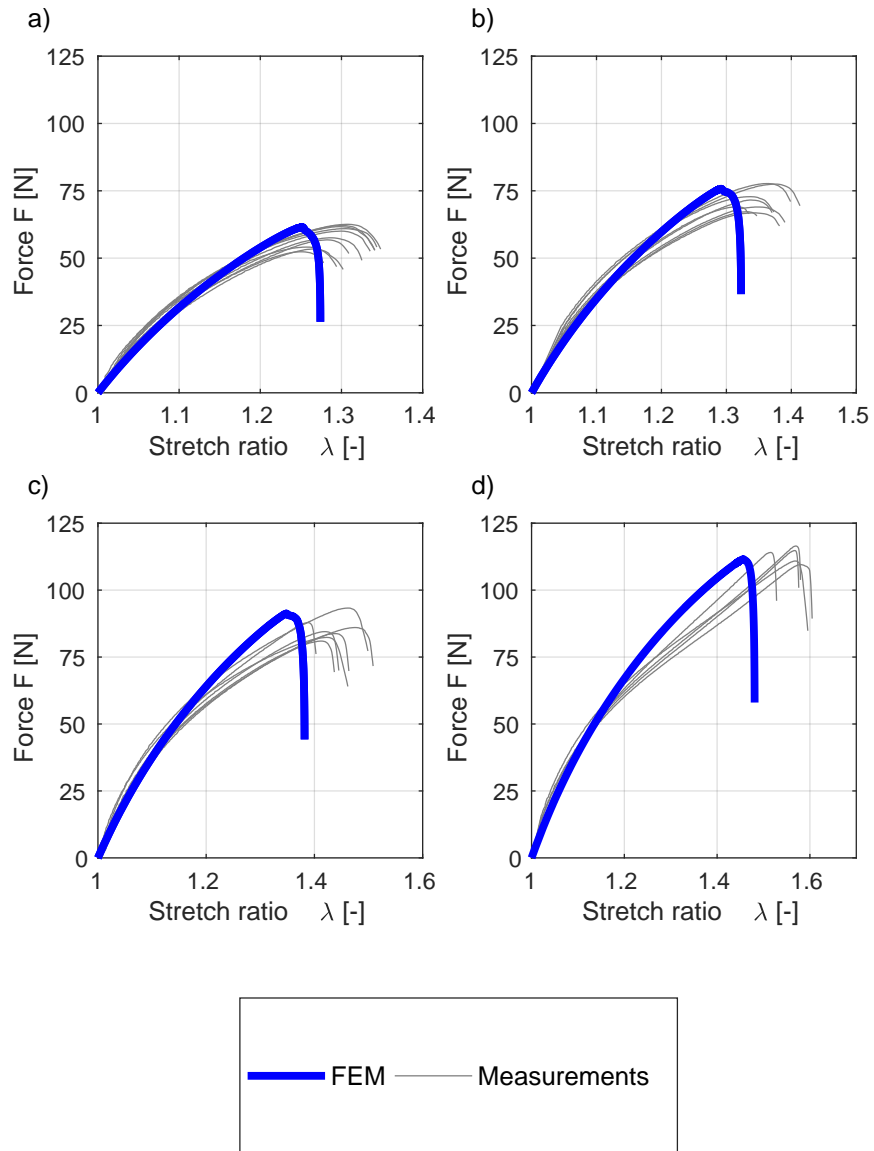


Figure 4.14: DENT test: Numerical and experimental force to stretch ratio response for a clamp velocity $75\text{mm}/\text{min}$. a) Initial crack length $z = 9\text{mm}$ b) Initial crack length $z = 7\text{mm}$ c) Initial crack length $z = 5\text{mm}$, d) Initial crack length $z = 3\text{mm}$.

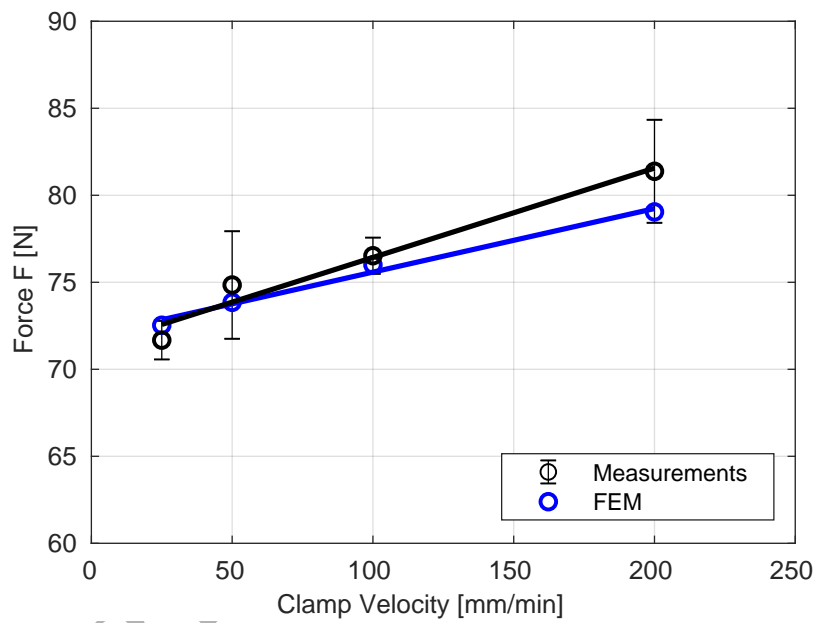


Figure 4.15: DENT tests: Numerically predicted and experimentally observed maximum tearing force (initial crack length $z = 7mm$) for different loading rates.

446 *4.7.3. Multi-notch tensile test*

447 The last geometrical set up includes three initial cracks (see figure 4.16), which
 448 coalesce during elongation. This example highlights the capabilities of the
 449 phase-field damage method to track complex crack patterns. Comparing fig-
 450 ures 4.16 a) and b), we see that the numerical predicted crack path matches
 451 with the one observed in the experiment. Furthermore, we see in figure 4.17 an
 452 acceptable match of the experimental force to stretch ratio responses.

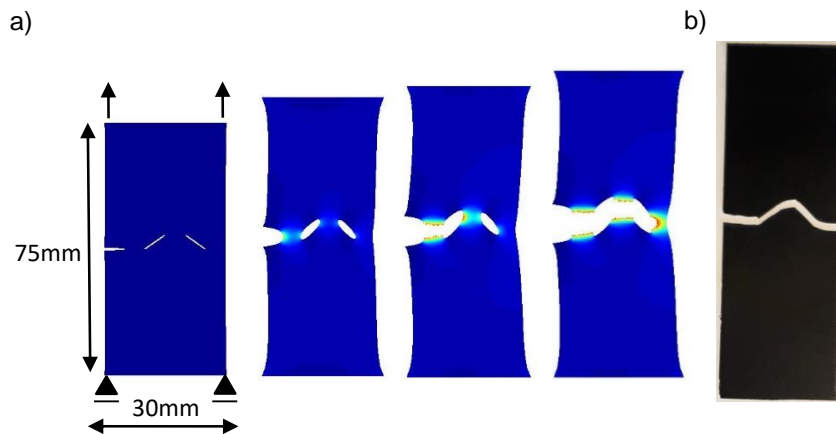


Figure 4.16: Multi notch tensile test: a) Numerically predicted crack path. b) Crack in the experiment.

453 *4.7.4. Double-edge notch creep tensile test*

454 Non-crystallizing rubbers such as EPDM show continuous crack growth under
 455 static loading. We conduct a double-edge notch tension creep test (initial crack
 456 length $z = 7mm$). A force of $65N$ is applied (clamp velocity $100mm/min$)
 457 leading to a displacement $u_{y,65N}$ at time $t = 0s$. The model, as can be seen in
 458 figure 4.18, predicts the time to failure accurately.

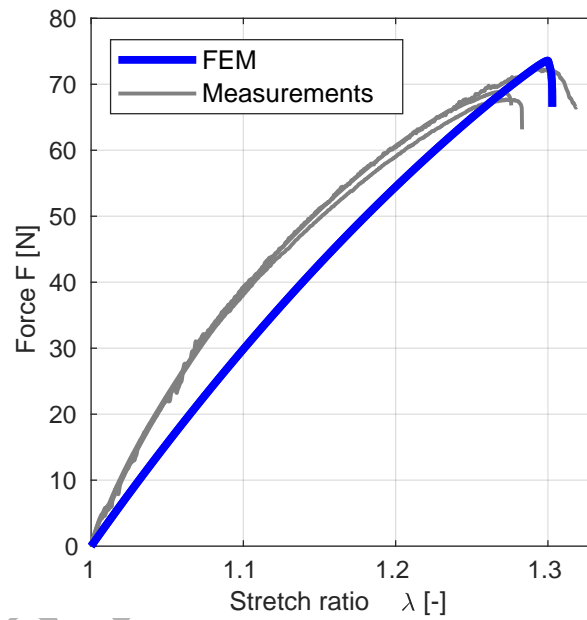


Figure 4.17: Multi notch tensile test: Numerically predicted and experimentally measured force to stretch ratio response, clamp velocity: $25\text{mm}/\text{min}$.

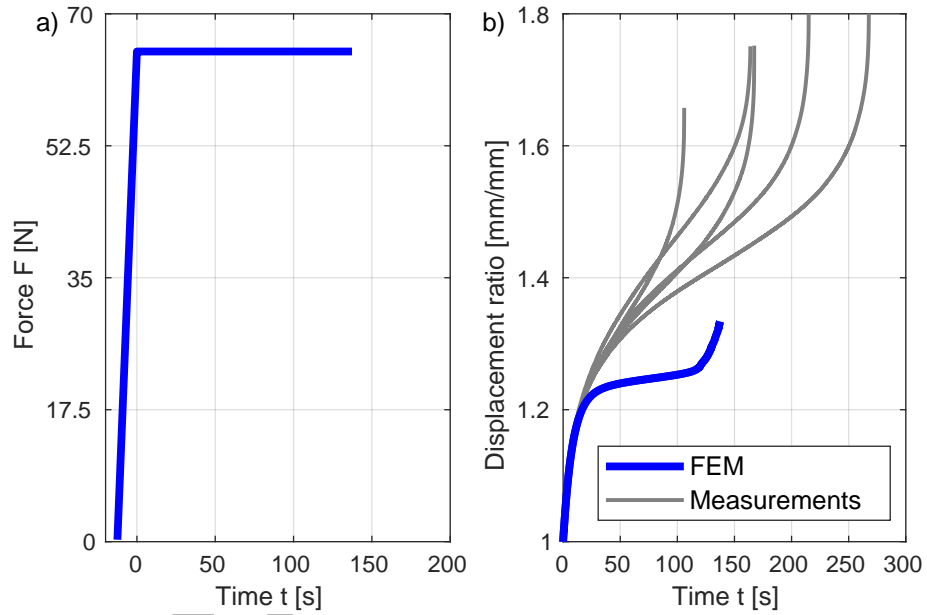


Figure 4.18: DENT creep test: a) Applied force over time b) Numerically predicted and experimentally measured displacement ratio $\frac{u_y}{u_{y,65N}}$ over time for a constant force of 65N.

459 5. Concluding remarks

460 A rate-dependent phase-field damage model is introduced. We have included
461 the rate-dependency in the damage formulation as well as in the constitutive
462 behavior of the bulk. The bulk response is modeled with a reduced polynomial
463 hyperelastic material model with three parameters [36] and the bulk's viscos-
464 ity is incorporated according to [37]. The material parameters for the bulk
465 are calibrated with uniaxial tensile tests, while the fracture parameters are ob-
466 tained from single and double-edge tensile tests with different clamp velocities.
467 Capturing local strains near the crack tip with digital image correlation has
468 allowed us to identify the length scale parameter. We have also assessed three
469 different degradation functions and have observed that the quadratic one fits
470 the experimental data best. The presented validation cases, which are true pre-
471 dictions, have shown amongst others that the model is capable to accurately
472 predict the time to failure for a creep test. Future work may extend the model
473 to incorporate temperature dependency and fatigue damage.

474 **Appendix A. Results degradation function $g_{d,2}$ and $g_{d,3}$**

475 In this appendix, we present the results for of the SENT test with an initial crack
 476 length $z = 20\text{mm}$, as depicted in figure 4.1a), for a loading rate of $25\text{mm}/\text{min}$
 477 and $200\text{mm}/\text{min}$. Subsequently, the results of the DENT test (figure 4.1b) for
 478 clamp velocity $75\text{mm}/\text{min}$ and varying crack size $z = [3\text{mm}; 5\text{mm}; 7\text{mm}; 9\text{mm}]$
 479 are plotted, as well as the results for the DENT test with fixed crack length $z =$
 480 7mm and varying clamp velocity. At first, we show the result for the degradation
 481 function $g_{d,2}$ (figure A.1, A.2 and A.3), then the results for degradation function
 482 $g_{d,3}$ (figure A.4, A.5 and A.6).

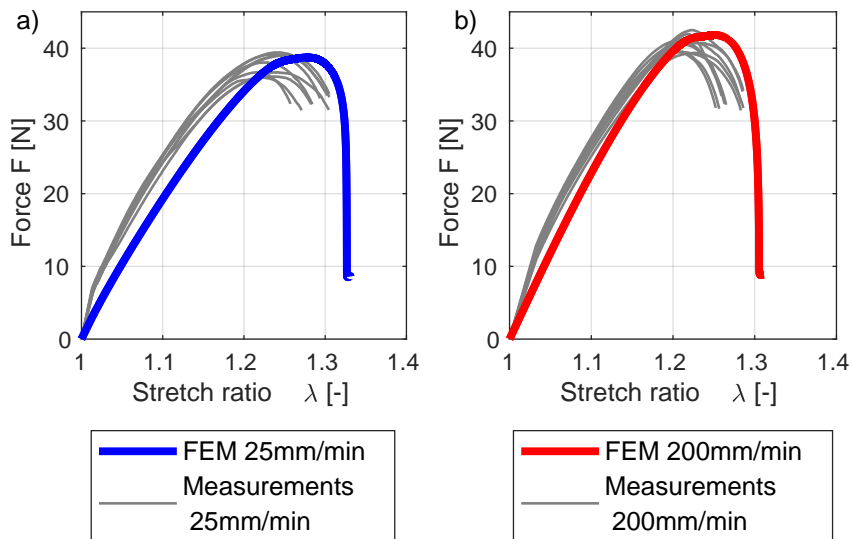


Figure A.1: SENT test for degradation function $g_{d,2}$: Numerical and experimental results
 a) Force to stretch-ratio response for $25\text{mm}/\text{min}$. b) Force to stretch-ratio response for
 $200\text{mm}/\text{min}$.

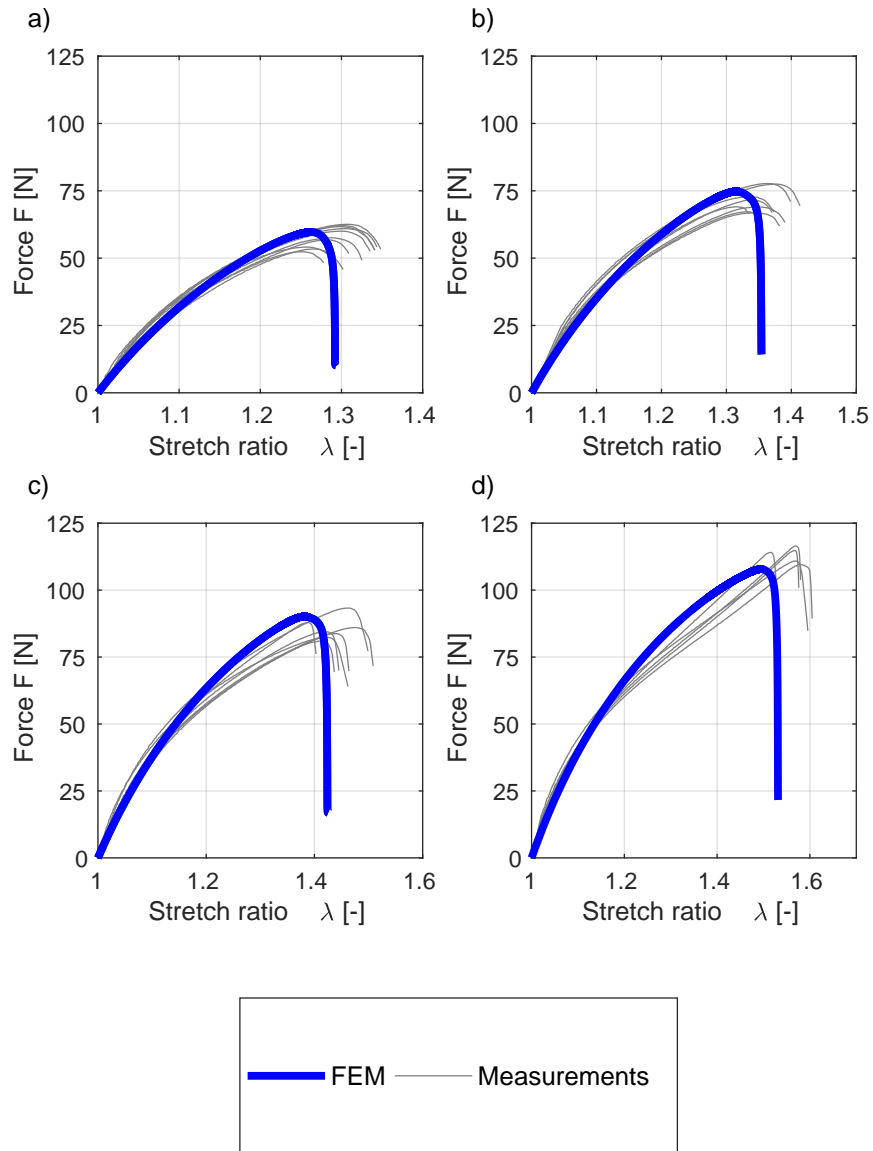


Figure A.2: DENT test for degradation function $g_{d,2}$: Numerical and experimental force to stretch ratio response for a clamp velocity $75\text{mm}/\text{min}$. a) Initial crack length $z = 9\text{mm}$ b) Initial crack length $z = 7\text{mm}$ c) Initial crack length $z = 5\text{mm}$, d) Initial crack length $z = 3\text{mm}$.

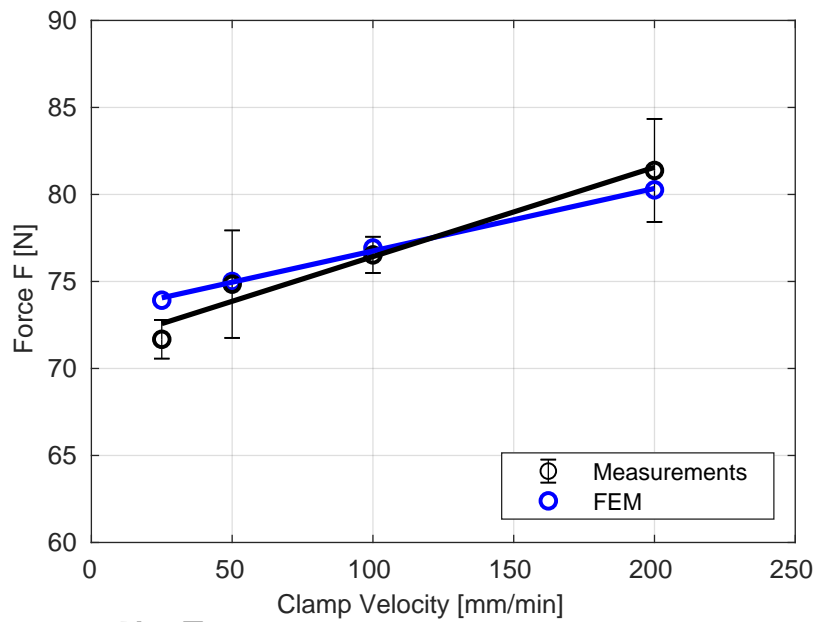


Figure A.3: DENT tests for degradation function $g_{d,2}$: Numerically predicted and experimentally observed maximum tearing force (initial crack length $z = 7mm$) for different loading rates.

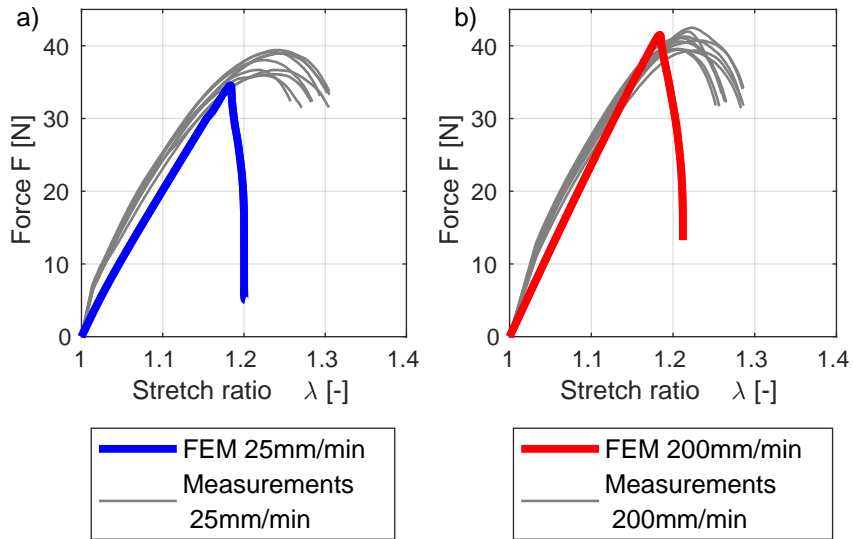


Figure A.4: SENT test for degradation function $g_{d,3}$: Numerical and experimental results
 a) Force to stretch-ratio response for $25\text{mm}/\text{min}$. b) Force to stretch-ratio response for $200\text{mm}/\text{min}$.

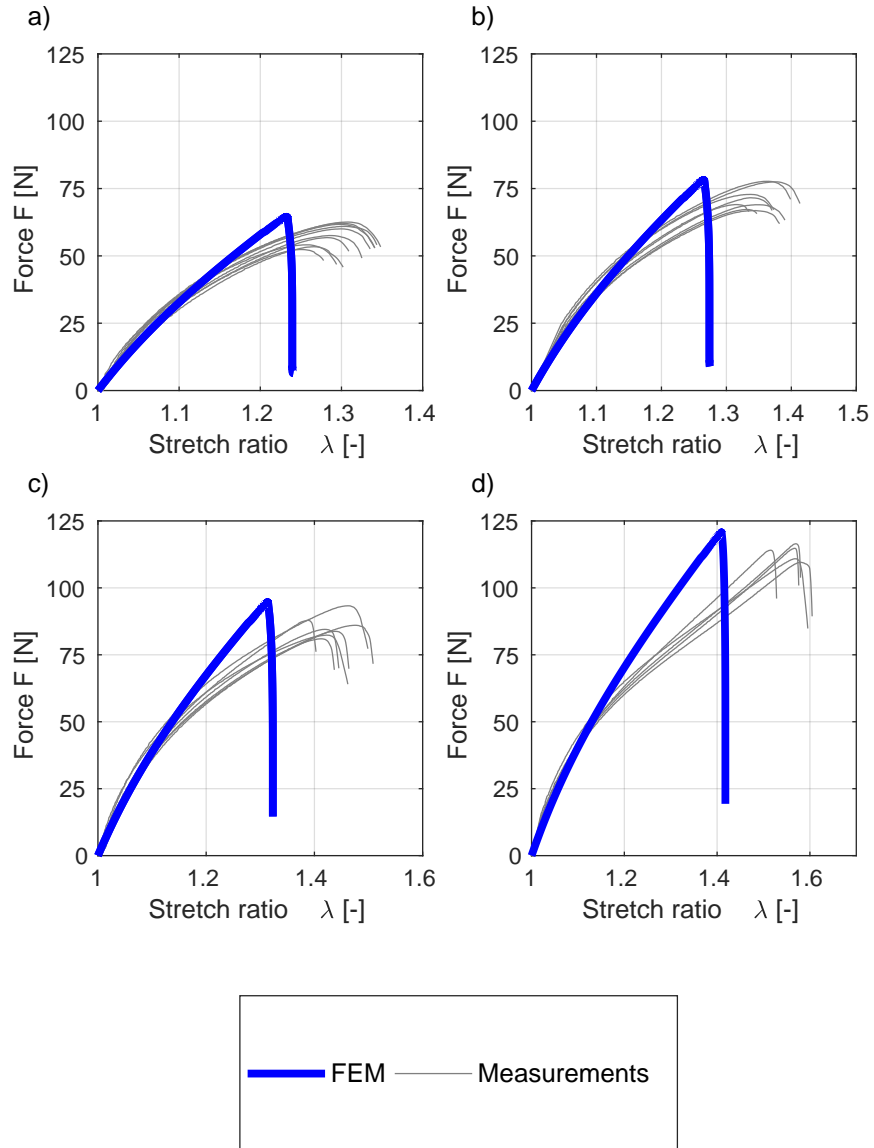


Figure A.5: DENT test for degradation function $g_{d,3}$: Numerical and experimental force to stretch ratio response for a clamp velocity $75\text{mm}/\text{min}$. a) Initial crack length $z = 9\text{mm}$ b) Initial crack length $z = 7\text{mm}$ c) Initial crack length $z = 5\text{mm}$, d) Initial crack length $z = 3\text{mm}$.

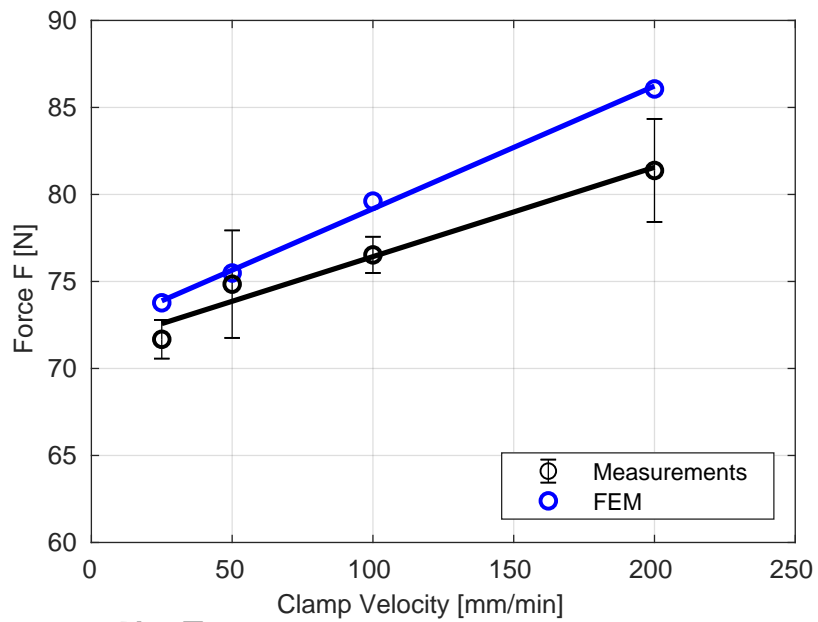


Figure A.6: DENT tests for degradation function $g_{d,3}$: Numerically predicted and experimentally observed maximum tearing force (initial crack length $z = 7mm$) for different loading rates.

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